

TECHNICAL REPORT  
HTL-TR-41

FINAL REPORT  
CONTRACT NAS8-11850  
Vol. IV  
TURBULENT MIXING OF  
THREE CONCENTRIC  
REACTING JETS

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September 29, 1967

Prepared For

National Aeronautics and Space Administration  
George C. Marshall Space Flight Center  
Huntsville, Alabama

## ACKNOWLEDGEMENT

The efforts of Mrs. E. M. McAllister and Mr. J. V. Hyden as the scientific programmers for this analysis are gratefully acknowledged.

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## PART II

### TURBULENT MIXING OF THREE CONCENTRIC REACTING JETS

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#### SUMMARY

The results of a computer program based on a modified P - function solution to the multiple-stream-turbulent-mixing problem are presented in graphical form. Though a computational problem persists in the determination of the absolute magnitude of the temperature, the plots of the normalized properties indicate that the trends of the mixing and combustion processes are properly defined. Recommendations to solve the computational problem in the temperature evaluation and to improve the program are outlined. It is concluded with the addition of a recommended subroutine the program would provide a method of describing mixing of equilibrium gases with greater speed than the available programs involving non-equilibrium calculations.

Complete listings of the two-stream-mixing program and the three-stream-mixing program are presented. The input format is illustrated by sample input sheets, for both two and three stream calculations.

## INTRODUCTION

A digital computer program has been constructed to evaluate the mixing and combustion characteristics of a rocket exhaust jet with turbine gases exhausting peripherally about the nozzle exit and with a surrounding external air stream. This study is a continuation of work reported in Part I (Ref. 1) of this report. Part I describes the analytical model and the equations used in the program. Computer runs were made for cases approximating the after-burning of H-1 engine flows at low altitudes. The output is in the form of velocity, temperature, and species mass fractions plotted as functions of position. The results of the three-stream-mixing analysis are presented.

## LIST OF SYMBOLS

$c_i$	Mass fraction of the $i$ th species
$H$	Total enthalpy, cal/g
$h$	Static enthalpy (intermediate value defined by Eq. (9), Ref. 1), cal/g
$h'$	Static enthalpy (defined by Eq. (10), Ref. 1) cal/g
$r$	Radius, cm
$T$	Temperature, °K
$U$	Velocity in the $x$ direction, cm/sec
$x$	Distance in axial direction, cm

Z       $\psi/\psi_e$   
 $\xi$     Transformation variable in axial direction (defined by Eq. (16), Ref. 1)  
 $\rho$     Density, g/cm<sup>3</sup>  
 $\psi$     Dimensionless stream function defined by Eq. (4), Ref. 1  
 $\bar{\psi}$    Dummy integration variable in the P-function integration

Subscripts:

e      Indicates external conditions  
j      Indicates rocket exhaust exit conditions  
ref    Reference values  
te     Indicates turbine exhaust exit conditions

Superscripts:

\*      Dimensionless quantity

## DISCUSSION

The solution of the flow equations for the velocity and energy values at a position in a transformed mathematical plane was obtained from the P function representation of the mixing process. With these physical conditions specified, the thermodynamic properties were then determined. All chemical processes were represented by one-step reactions. An atom balance was used with molecular composition determined to comply with the equilibrium restraint. Transformations were used to position these results in the physical plane.

Program Input. - The program constructs and solves the equations defining the mixing and combustion according to the numerical (key) instructions placed in the input. A prior knowledge of the possible reaction products is required. A lower limit upon the mass fractions considered is used as an input variable to allow flexibility in the compromise between accuracy desired and required run time.

The P function solution for the three-stream-mixing cases is modified using a table look-up of the modification factor. The following equation

$$\text{Modification Factor} = \frac{\left( \begin{array}{l} \text{Value at Point} \\ \text{of Consideration} \end{array} \right) - \left( \begin{array}{l} \text{Value in the} \\ \text{External Stream} \end{array} \right)}{\left( \begin{array}{l} \text{Rocket Exhaust} \\ \text{Value} \end{array} \right) - \left( \begin{array}{l} \text{External Stream} \\ \text{Value} \end{array} \right)}$$

was used to determine this factor from the results of the two separate two-stream mixing cases. The modification factor is evaluated for each elemental composition, the enthalpy and the speed. With these modification terms in the calculation procedure, it is possible to represent arbitrary additions of energy and chemical composition such as that of the turbine exhaust. The modified P function solution is still a similar type solution with a Gaussian decay but the initial profiles may be of arbitrary shape.

Because of the equilibrium restriction the program is highly dependent upon the input thermodynamic data. Equilibrium constants were input directly from Ref. 2 in tabular

form. Thermodynamic constants from Ref. 3 were input in array format for use in evaluating the static enthalpy in the mixture of gases present.

Program Internal Structure. - The P-function solution was modified by a factor obtained by normalizing the conditions present at the sub-initial plane. In the three-stream-mixing problem, the normalized conditions at the sub-initial plane replace the initial values used previously in the two-stream mixing case. To provide for the addition of arbitrary chemical species and energy levels, each of the properties and elements are calculated separately using the appropriate modification factor determined by the normalization procedure. A table look-up of the modification factor with provisions for linear interpolation of values is used. This method of storing and retrieving the modification terms was found to be faster and more accurate than a polynomial representation of the modifying factor in the P-function.

The thermodynamic condition of the equilibrium mixture is determined by constructing a matrix of equations involving the unknown mass fractions, density, and temperature. The equations are a combination of algebraic and transcendental equations. A Newton-Raphson iteration technique is applied to evaluate the unknowns. Logarithmic corrections for the gaseous constituents, as shown in Ref. 4, are used to accelerate the convergence of the equilibrium composition calculations.

The atom balance is maintained by use of a numerical

code for each molecule within the system. In the code each atomic species is assigned a particular numerical designation.

Separate subroutines are provided for the transformation of the results from the mathematical plane to the physical plane. The transformation to the axial and the radial position locations are independent subroutines so that the uses may easily alter the form of either of the transformations.

Computational Problems. - Some computational problems persist in the program. Iteration is carried out upon the static enthalpy of the mixture to determine the correct temperature. This involves the calculation of the partial enthalpy for each species from the calculated values of temperature and the mass fractions. To limit computer run time the number of iterations necessary to determine the density and mass fractions must be kept low. This was accomplished by adjusting the tolerance on the mass fraction calculation. Investigation of several cases shows that the mass fractions of some species, though present only in trace amounts, influenced the calculated enthalpy value of the mixture because of their large heat of formation values. The temperature of the mixture is calculated from the enthalpy iteration. Because of the influence of the species in trace amounts with high heats of formation, the computed magnitudes of the temperature are too high. Since the temperature plots appear as normalized curves, the trends and profiles are nevertheless properly described. A possible solution to this problem of

computing the temperature is to establish a test for each species mass fraction and to eliminate those that appear only in trace amounts from the enthalpy calculation. Experimentation would be required to determine the proper limit to place upon the mass fractions. There was insufficient time to attempt any trial of this approach.

Divergence of the solution for the mass fractions and density occurs when the initial estimates are far from the correct values. This is an inherent problem of the Newton-Raphson iteration procedure (Ref. 5). Methods to speed or force convergence exist but it was considered unadvisable to resort to these more sophisticated methods while problems such as that with the enthalpy iteration were present.

A study of expedient methods to eliminate these problems indicated that the addition of the capability of using the Lewis Thermochemical program (Ref. 3) as a subroutine would be the most useful. Inclusion of this capability would not only solve the two previously described problems but in addition would make it unnecessary to provide a priori information about the resultant products of the reactions.

The addition of this subroutine would be facilitated by using the modification of the Lewis program carried out by R. Taylor (Ref. 6) to calculate adiabatic flame temperatures. It presently uses an internal table look-up for variable fuel to oxidant ratios. This modification is restricted at present to a minimum of two fuel to oxidant

ratios. With this restraint removed, the subroutine could be used to calculate the thermodynamic properties given the present input to the Newton-Raphson subroutine.

Program Output. - The output of the program is printed in increments of the transformed stream function for a fixed transformed axial location. Values for the mass fractions, temperature, density, velocity, and enthalpy are printed after each increment. These values are then located in the physical plane with a print-out of the corresponding axial and radial positions.

Presentation of Results. - The results of the analysis of the mixing and combustion are shown as plots in Figs. (1) - (11). The values are plotted as normalized values using initial conditions of the rocket exhaust exit, the turbine exhaust, or the external stream as indicated in the individual coordinate labels. The normalized values are plotted versus the radial location for the fixed axial location which corresponds to the fixed transformed axial location.

Two separate mixing zones were calculated for two-stream-mixing. One case represented the mixing of rocket exhaust and turbine exhaust while the second was for the mixing of turbine exhaust and the external air stream. The intersection of these two mixing zones was calculated to occur at a radial position of 23.3 inches and an axial location of 1.2 inches from the nozzle exit. The sub-initial plane was located by

this intersection.

Examination of the trends shown by the normalized temperature plots indicate that mixing and combustion of the relative cold turbine exhaust and rocket exhaust are properly described. The burning of the exterior portion of the turbine exhaust is shown. The cooling of the rocket exhaust by the cooler turbine exhaust mixing together on the periphery of the rocket exhaust is shown as a dip in the curves in Fig. (2). The onset of combustion of the turbine exhaust is displayed by the rise in temperature following the dip in the temperature curve.

The maximum peak of the temperature curves indicates the secondary combustion of the fuel rich rocket exhaust and turbine exhaust mixture which has mixed with the surrounding air. It should be noted that the temperature caused by the combustion of the turbine exhaust products is less than the initial temperature value of rocket exhaust.

The plot of normalized mass fraction of water (Fig. (6)) displays the action of mixing of the turbine exhaust and rocket exhaust both being fuel-rich with no combustion occurring. This is indicated by the dip in the water profile. As a fuel rich turbine exhaust mixes with the surrounding air, combustion produces a rapid rise in the mass fraction of water present.

The remaining species reflect the mixing and combustion as shown in Figs. (4) - (11).

## CONCLUSIONS AND RECOMMENDATIONS

Examination of the plots of normalized properties of the three-mixing-streams indicates that the trends of the process are properly defined. A computational problem of determining the correct magnitude of temperature still persists because of the dominating influence in the enthalpy iteration of species with high heats of formation which are in reality present only in trace amounts. Though the absolute magnitude of the temperature curves is too high, the pattern of the normalized temperature predicts the cooling of the rocket exhaust through mixing with the turbine exhaust and the rise in temperature because of combustion of the mixture of turbine exhaust and air as the radial position moves outward from the centerline. The mixing without combustion of the fuel-rich rocket exhaust and turbine exhaust followed by the combustion of the fuel-rich turbine exhaust when mixed with the surrounding air is also predicted by the plots of the normalized mass fractions of the species.

It is recommended that capability of using the Lewis Thermochemical program as a subroutine be added to the present program. This would eliminate the need for a priori knowledge of reaction products and by-pass the Newton-Raphson subroutine used to calculate the unknown mass fractions.

It is concluded based upon the normalized profiles that the modified P-function adaptation of the similar profile

method is a feasible approach to the multiple-stream-mixing problem. With the addition of the recommended subroutine the program would provide a method of describing mixing of equilibrium gases with greater speed than the available programs involving non-equilibrium calculations.

## APPENDIX A

### COMPUTER PROGRAM

The following description provides information for those desiring the particular results formulated by the program. This program deals with symbols and is very flexible in the number of equations and species which may be used as known information. Listings of both the two and three-stream-mixing programs are given. A sample is presented for the two-stream-mixing in Figs. (12) - (15). Three-stream-mixing sample input is shown in Figs. (16) - (21).

Limitations. - The program has the following limitations:

1. The maximum number of input equations is 10.
2. The maximum number of species is 19.
3. Each specie table has a maximum of 20 values.
4. The maximum number of equations that can be formed from the input equations is 15.
5. The initial estimate for each unknown species must be positive.
6. The values of the species tables must be input as logarithms of the base ten. This is really not a limitation since most thermochemical tables are given in this form.

List of Input Variables. -

1. KT = number of values in each species table
2. NSMAX = number of species in the equations omitting nitrogen

3. NRO = number of input equations
  - a. NRO1 = NRO + 1
  - b. NRO2 = NRO + 2
  - c. NSMAX1 = NSMAX + 1
  - d. NSMAX2 = NSMAX + 2
  - e. NRS = NSMAX1 - NRO1
  - f. NRS1 = NRS + 1
4. NXG, a zero indicates one set of initial estimates of the unknown species mass fractions; a one indicates multiple sets of initial estimates of the unknown species mass fractions.
5. Dl(I) where I = 1, NRS1 are constants used in forming the last NRS1 equations. It is the mass fraction of elements in the outer mixing region.
6. Gl(I) where I = 1, NRS1 are constants used in forming the last NRS1 equations. It is the mass fraction of elements in the inner mixing region.
7. SMASS (I) = mass of species I where I = 1, NSMAX
8. NALPHP (I,J) where I = 1, NRO and J = 1, NSMAX (see below)
9. NALPHPP (I,J) where I = 1, NRO and J = 1, NSMAX (see below)  
NALPHP and NALPHPP will be discussed later.
10. PSI =  $\psi$
11. DPSIBAR =  $\Delta\bar{\psi} = \Delta Z$
12. DXI =  $\Delta\xi$
13. DPSI =  $\Delta\psi$
14. XI =  $\xi$
15. PSIMAX = maximum value of  $\psi$
16. XIMAX = maximum value of  $\xi$
17. TREF =  $T_{ref}$ , Reference temperature
18. PREF =  $\rho_{ref}$ , Reference density
19. REFV = Reference number of moles

20. HE =  $H_e$
21. HJ =  $H_j$
22. ACCTOL = tolerance of  $|h-h'|$  used in the enthalpy iteration to determine temperature
23. PSIJ =  $\psi_j$
24. T = T is the input estimate for the temperature
25. DELTAT =  $\Delta T$  or change in calculating a new T
26. DT =  $\Delta T$  used as dT in the integration to determine static enthalpy
27. TOL = tolerance setting for solution of simultaneous equations
28. PSIBAR =  $\bar{\psi} = Z$
29. UJ =  $U_j$
30. UE =  $U_e$
31. USTARJ =  $U_j^*$
32. RHOE =  $\rho_e$
33. RH =  $r_j$
34. CNITROM = mass of nitrogen
35. AACON (I,J,K) where I = 1,6 and J = 1, NSMAX with J = index of specie used for the thermochemical constants and K = 1,2 for higher and lower temperature ranges respectively.
36. A0 =  $a_0$  and PLYC =  $a_1, a_2 \dots, a_{20}$  are coefficients of the polynomial for  $F(\psi)$  used in the P-function integration. [Note: Used only in two-stream-mixing program.]
37. PSI TABLE is used for interpolation base for the modification factor tables. [Note: Used only in three-stream-mixing program.]
38. MODIFICATION FACTOR tables are of the same size as the PSI TABLE. There are complete tables for velocity, first element, second element, ..., thru NRS1 elements, and total enthalpy. [Note: Used only in three-stream-mixing program.]

39. TTAB(K) where K = 1,KT and used for interpolation base for T and species tables.
40. STAB (I,J) where I = 1,KT and J = 1, NSMAX with J = index of species. Therefore, there are NSMAX species tables with KT values in each.
41. AAHOT (I,K) where I = 1,6 and K = 1,2  
AAHOT is thermochemical constants for nitrogen.
42. KSPEC (I,J) where I = 1,4 and J = 1, NSMAX will be discussed later.
43. XG(I) are the estimates of the final values of the unknown species mass fraction where I = 1, NSMAX.
44. XG(I) CUT-OFF places a lower limit on the size of the species mass fractions to eliminate computer underflow.
45. SUM provides a means of restarting the calculation and still maintain the accuracy in the radial position integration. It is equal to the value of RSTARS print-out for the previous position.

Data Format. - Data is read by two different formats.

The first format reads fixed point integers by the following set up:

- a. Each data card has 12 fields of width 5. Each value must be right justified in the field. For example: If the numbers 20,5 and 2 are to be read then 20 would be placed in the first field, 5 in the second field and 2 in the third field. A 2 would be placed in column 4, 0 in column 5, 5 in column 10 and a 2 in column 15.
- b. If an array N has a dimension of 15 then the values N(1), N(2),..., N(12) would be punched in the first card with N(13), N(14), N(15) in the second card.
- c. If an array, say K, is two-dimensional of maximum dimension of 15 x 20 but only a 3 x 4 matrix is needed then 12 values would be read by columns in the order K(1,1), K(2,1), K(3,1), K(1,2), K(2,2), K(3,2),...,K(1,4), K(2,4), K(3,4), K(4,4).

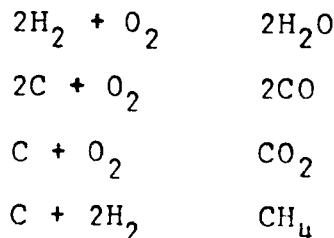
Note: Most of the input to this program is of the form A(I,J) with I = 1,..., NRO and J = 1,..., NSMAX where NRO and NSMAX are values which are known and change with each new case.

The second format reads real or floating point numbers by the following set up:

- a. Each data card has 6 fields of width 12. Each value may be input as a number which contains a decimal or as a number which is given with an exponent. For example; -.000560511 and -5.605011E-4.

Formulation of Input. NALPHAP(I,J) and NALPHPP(I,J) are two-dimensional arrays where I = 1,2..., NRO and J = 1,2,..., NSMAX. NALPHP and NALPHPP, the stoichiometric coefficients, are used in the program and must be set up in a particular way. These two arrays are used to input the coefficients of the reaction equations.

Example:



4 equations hence NRO = 2

7 species hence NSMAX = 7

Therefore, by reading the first equation from the left, there is H<sub>2</sub> (species 1), O<sub>2</sub> (species 2), and H<sub>2</sub>O (species 3). Reading through the remaining three equations gives C (species 4), CO (species 5), CO<sub>2</sub> (species 6), and CH<sub>4</sub> (species 7).

The input for NALPHP and NALPHPP is as follows:

NALPHP

H <sub>2</sub>	O <sub>2</sub>	H <sub>2</sub> O	C	CO	CO <sub>2</sub>	CH <sub>4</sub>
2	1	0	0	0	0	0

NALPHP (Con't)

H <sub>2</sub>	O <sub>2</sub>	H <sub>2</sub> O	C	CO	CO <sub>2</sub>	CH <sub>4</sub>
0	1	0	2	0	0	0
0	1	0	1	0	0	0
2	0	0	1	0	0	0

NALPHPP

H <sub>2</sub>	O <sub>2</sub>	H <sub>2</sub> O	C	CO	CO <sub>2</sub>	CH <sub>4</sub>
0	0	2	0	0	0	0
0	0	0	0	2	0	0
0	0	0	0	0	1	0
0	0	0	0	0	0	1

Notice that the column i corresponds to the coefficients of species i and row j corresponds to the equation j. For example, NALPHPP (2,5) represents the second equation and the fifth species (CO). The stoichiometric coefficient of CO in the second equation is NALPHPP (2,5) which is 2. When these arrays are loaded into the machine, all the NALPHP coefficients for the first species are inputed before the coefficients of the remaining species [See Fig. (12)].

KSPEC (I,J) is a two-dimensional array used in the program where I = 1,2,3,4, and J = 1,2,..., NSMAX with J = species J. For the above example,

KSPEC

H <sub>2</sub>	O <sub>2</sub>	H <sub>2</sub> O	C	CO	CO <sub>2</sub>	CH <sub>4</sub>
1	2	1	0	2	2	1
0	0	2	4	4	0	4
0	0	0	0	0	4	0
0	0	0	0	0	0	0

The species 3, 5, 6, and 7 are products of species 1, 2, and 4 hence  $H_2O$  is stored as a 1 in the first row indicating that it contains hydrogen,  $H_2$ , and a 2 in the second row indicating that it contains oxygen,  $O_2$ . It is important that the numerical key for the species appear in its proper column but it may be placed at random in the column. These values are input so that all of the first species column (even zeros) are input before the remaining species columns are input in order.

The D1(I) and G1(I) are tied to the species numerical key described in the NALPHP and NALPHPP arrays. From this D1(1) is the mass fraction of the element hydrogen, H, in the outer mixing region. D1(2) is the mass fraction of oxygen, O, in the outer mixing region. The last D1(I) value is the mass fraction of nitrogen,  $N_2$ . The G1(I) values use the same numerical keys and indicate the mass fraction of the elements in the inner mixing region.

# LISTING OF TWO-STREAM-MIXING COMPUTER PROGRAM

3200 FORTRAN (Z.1.0)/(RIS)

```

C      SUBROUTINE DEX
C      A0=COEFFICIENT OF THE POLYNOMIAL
C      PLYC=EFFICIENT OF POLYNOMIAL (NS1BAR)
C      AA0U(I,J,K)=CONSTANT USED IN FINDING HOT AND COLD
C      AA0U(I,J,K)=HOT TEMPERATURE RANGE FOR K=2 AND
C      AA0U(I,J,K)=COLD TEMPERATURE RANGE FOR K=1 AND
C      K=I=1,...,N AND J=1,...,K FOR SPECIE K.
C      AAH(I,J,K) WHERE J=1 OR 2 WITH J=2 IS FOR
C      THE COLD TEMPERATURE RANGE FOR NITROGEN.
C      TABS=TABLE OF 1
C      KNEW = 0 VALUES IN EACH SPECIE TABLE
C      TABL(I,J) ARE TABLES OF SPECIE VALUES
C      CORRESPONDING TO SPECIE J WHERE I=1,...,NI
C      I IS VALUE OF INPUT FOR TABLE L0K=J
C      AAUH,NLFP,HAA2,NSMAX,VR AND VNP ARE USED
C      TO SOLVE INPUT EQUATIONS
C      LI=INCREMENT FOR INTEGRATION
C      LELTATE=LICHEN FOR I ITERATION
C      L1(I) AND R1(I) FOR EQUATION I+NR0+1 AND C1(I)
C      CBAR(I) IS SOLUTION OF NON-LINEAR SYSTEM OF EQUATIONS
C      CHAS=(I) IS CONCENTRATION OR MASS OF SPECIES I
C      UNITHD IS USED IN NR0+1 EQUATION
C      UNITHM=ASSUMED MASS OF NITROGEN
C      NNEW = 1 CAUSES NEW CASE TO BE READ,
C      NNEW IS SET IN NEWTON.
C      XG=1 IMPLIES THAT NO XG GUESSES ARE AFTER CASE.
C      XG=0 IMPLIES THAT XG GUESSES ARE AFTER CASE.
C      XGMIN IS THE SMALLEST VALUE THAT
C      XG CAN TAKE IN SECTION (1,F1) VALUE)
C
C      DIMIN XU,I,N,I,J,AHA
C      DIMIN TAB(20),SIAM(20,20)
C      DIMENSION NSPEC(4,20)
C      DIMENSION PLYC(20),KELEM(20),AA0U(6,20,2),AAH(6,2)
C      DIMENSION HAA(10,20),VAA(10,20),VR(15),VNP(10)
C      DIMENSION NLFP(10,20),NLFPH(10,20)
C      DIMENSION FRC(10),FRL(10),FAB(20),ARRAY(1,22)
C      DIMENSION XG(20),CEAR(21)
C      DIMENSION CHAS(20),E1(15),G1(15)
C      DIMENSION HUM(14,21),ULX(21),ARRAY(21,22)
C      DIMENSION SSTAT(20,20)
C      EQUIVALENCE (FRC,I),(FRL,I,F2)
C
C      READ 1 PUT
C
C      1  READ
C      1000 READ LIN,NSXT,NSMAX,NR0,XG
C      I1=1
C      I10=10
C      NR1=NR0+1
C      NR2=NR0+2
C      NS=A+1=NSMAX+1
C      AS=A+2=NSMAX+2
C      NRSE=NS-A+1-NR0+1
C      NRSL=NS+1
C      NRFA=0

```

```

      READ (IN,3)(H1(1),I=1, NS1)
      READ (IN,3)(L1(1),I=1, NS1)
      READ (IN,3)(S,AS5(1),I=1, NSMAX)
      READ (IN,2)((ALPHF(I,J),I=1,NR0),J=1,NSMAX)
      READ (IN,2)((LRHF(I,J),I=1,NR0),J=1,NSMAX)
      READ (IN,3)PS1,PS1,LHOBAR,X1,DX1,PSIMAX,XIMAX,IREF,FREF,RREF,
      1, R, H, A001, L, PS1, I, LE, TA1, DL, TUL, PHBAR, UU, UB, USTAR0, RH00, R00
      C
      READ (IN,3)((AACUN(I,J,1),I=1,6),J=1,NSMAX)
      READ (IN,3)((AACU,(I,J,2),I=1,6),J=1,NSMAX)
      READ (IN,3)((AAHO((I,J),I=1,6),J=1,2)
      READ (IN,3)AL,PLYC
      READ (IN,3)(TA8(K),K=1,K1)
      READ (IN,3)((STAR(I,J),I=1,KT),J=1,NSMAX)
      READ (IN,2)((NSREC(I,J),I=1,4),J=1,NSMAX)
1002 READ (IN,3)(XU(I),I=1,NSMAX)
      READ (IN,3)XMIN
      1 READ(1,NSUM
      3 FORMAT (6F12.2)
      2 FORMAT (1Z15)

C
C   SET UP NAA AND NAA2 ARRAYS FROM NALPHF AND NALCHPF
C
      DO 470 I=1,NHU
      DO 470 J=1,NSMAX
      NAA(I,J)=0
      NAA2(I,J)=0
470  DO 471 E
      DO 470 I=1,NHU
      DO 471 J=1,NSMAX
      IF ((ALPHF(I,J))452,453,452
452  NAA(I,J)=J
453  IF ((LRHF(I,J))454,451,454
454  NAA2(I,J)=J
451  CONTINUE
450  CONTINUE

C
C   SET UP NF AND NNP ARRAYS FROM NALPHF AND NALCHPF
C
      K=1
      J=1
456  J=J+1
      DO 457 I=1,NHU
      IF ((ALPHF(I,J))455,455,458
458  F(K)=
      K=K+1
      IF ((J-K))456,456,456
455  DO 456 E
      IF ((J-K))456,456,456
459  CONTINUE
      K=1
      J=1
466  J=J+1
      DO 467 I=1,NHU
      IF ((LRHF(I,J))465,465,468
468  F(K)=J
      K=K+1
      IF ((K-K))466,466,466
465  CONTINUE
      IF ((K-K))466,466,466
469  CONTINUE
      C

```

```

C      PRINT INPUT
C
      WRITE(11,41)N1,...MAX, R
  41  FORMAT (1H),//,/RHEC MIXING SUB NO. 560180//,
     151H 0.00000E+00 IS 1, TABLES =13,4X,
     220H 0.00000E+00 IS 1, EQUATIONS =13,4X,
     327H 0.00000E+00 IS 1, EQUATIONS =13/
     WRITE(11,42)(I1(I),I=1,VS1)
  42  FORMAT (0H,01(I)/,(5E22.9))
     WRITE(11,43)(B1(I),I=1,VS1)
  43  FORMAT (0H,B1(I)/,(5E22.9))
  44  FORMAT (1H, DATA OF SPECIES 1/, (5E22.9))
     WRITE(11,45)(XG(I),I=1,VSMAX)
  45  FORMAT (1H, DATA OF SPECIES 2/, (5E22.9))
     WRITE(11,352)XGMIN
  352  FORMAT (/>1/H XG & I</> M VALUES,DX=17.9//)
     WRITE(11,46)((VALFHE(I,J),I=1,NH),J=1,VSMAX)
  46  FORMAT (/>, /> NALFHE/, (10!10))
     WRITE(11,47)((VALFHF(I,J),I=1,NHF),J=1,VSMAX)
  47  FORMAT (FH VALFHF/, (10!10))
     WRITE(11,48)((VAA(I,J),I=1,NR),J=1,VSMAX)
  48  FORMAT (FH VAA(I,J)/,(10!10))
     WRITE(11,49)((VAA2(I,J),I=1,NR),J=1,VSMAX)
  49  FORMAT ((10!10)VAA2(I,J)/,(10!10))
     WRITE(11,50)(NH(I),I=1,NH)
  50  FORMAT (FH NH(I)/,(10!10))
  51  FORMAT (7H VEF(I)/,(10!10))
     WRITE(11,51)(NAP(I),I=1,NR)
  52  FORMAT (7H VEF(I)/,(10!10))
     WRITE(11,52)PS1,PS1,PSHAR,X1,DX1,PSIMAX,X1,DX1,
     1HRR,PFET,PFET,V,PF,PAOT,PF,PAOT,PF,PAOT,PF,PAOT,
     2FHSA,PAOT,PAOT,PAOT,PAOT,PAOT,PAOT,PAOT,PAOT
  52  FORMAT (//,6H PS1 E17.9,4X,5H 0PS1E17.9,2X,7H E17.9,
     14X,5X E17.9,4X,5H E17.9,6H E17.9,4X,5H E17.9,4X,
     2DHX1 AXE17.9,4X,5H HEE E17.9,4X,5H HEE E17.9,4X,
     3DHHEEV E17.9,4X,5H HE E17.9,4X,5H HEE E17.9,3X,
     46HALU1LF17.9,4X,5H S1J E17.9,4X,5H S1J E17.9,7,
     56H 0.0LATE17.9,4X,5H BLT E17.9,4X,5H BLT E17.9,
     13X,6H P1E17.9,4X,5H HJ E17.9,6H UE E17.9,
     73X,6H UTAH E17.9,4X,5H HHDJ E17.9,4X,5H HHDJ E17.9,3X,
     64H E17.9,4X,5H HHDJ E17.9,4X,5H HHDJ E17.9,4X,5H HHDJ E17.9)
  53  FORMAT (10!4// K=1,2
     10! 431 J=1,VSMAX
     WRITE(11,432)J,(AAHCN(I,J,K),I=1,6)
  432  FORMAT (2D8.1,15H AAHCN(I,J,K) J=12/, (5E22.9))
  431  FORMAT (10!4// E
     1H (r-2)4/3,4/2,4/2
  473  WRITE(11,474)
  474  FORMAT (//,19X,35H AAC N ARRAY FOR LOWER TEMPERATURES/)
  475  FORMAT (10!4// E
     1H (r-2)4/3,4/2,4/2
  476  FORMAT (7,20X,5HAAHT/,(5E22.9))
     WRITE(11,477)(AAHT(I),I=1,6)
  477  FORMAT (//,20X,28HAAHT1 FOR LOWER TEMPERATURES/, (5E22.9))
     WRITE(11,478)AO,PLYC
  478  FORMAT (//,19X,13H AC,A1,...,AN/, (5E22.9))
  84  FORMAT (//,19X,13H AC,A1,...,AN/, (5E22.9))
C      PRINT TABLES
C
      WRITE(11,493)(TAPE(K),K=11,17)

```

```

193 FORMAT (//,20X,7HT TABLE//,(5E22.9))
      WRITE (10,444)
444  FORMAT (//,20X,32HSPECIE TABLES WITH X=NUMBER//)
      DO 260 I=1,NSMAX
      WRITE (10,194) I
194  FORMAT (20X,9HSPECIES =13,6H TABLE)
      WRITE (10,195)(STAR(K,I),K=1,K1)
195  FORMAT (5E22.9)
200  GO TO 196
C
C      SETI=00  RETEN(1)=1
C
C      DO 2 K=1,20
C          RETEN(K)=K
      5  GO TO 196
C
C      DO 350 N=1,5148,...,520148 TABLES FROM X=00(Y) TO Y VALUES
C          IF (N=1)100,350,100
100  IF (N>6)350,350,351
350  NFA=1
      DO 250 J=1,K1
      DO 250 I=1,NSMAX
255  STAR(I,J)=STAR(1,J)
C
      351 DO 260 J=1,NSMAX
          I=260 I=1,K1
          IF (TAB(1,J))493,491,491
491  TAB(1,J)=10.0***TAB(1,J)
          GO 1 490
493  TAB(1,J)
      STE=1
      RETEN(ST)
      FST=AHS(ST)-AHS(FNST)
      TAB(1,J)=10.0***FST+10.0***FNST
490  GO 196
260  GO TO 196
C
      WRITE (10,445)
445  FORMAT (//,20X,27HSPECIE TABLES WITH X=NUMBER//)
C
      DO 401 J=1,NSMAX
      WRITE (10,402) J,(STAR(I,J),I=1,K1)
402  FORMAT (20X,13HSPECIE TABLE 12/,5E22.9)
401  GO TO 196
      WRITE (10,219)
219  FORMAT (//,20X,24HSPEC(1,J) WHERE J=INDEX//)
      DO 89 I=1,NSMAX
      WRITE (10,89) (SPEC(I,J),J=1,4)
89   GO TO 196
89   FORMAT (3X,41D)
C
C
C      READ IN VALUES
C
      EPS1=0
      DO 100 I=1,L
      READ(ADEPSBAR,EPS1)
      EPS1=EPS1
      T0=300.
      N=1,167

```

```

C      400 IS THE RETURN POINT FOR PST AND
C      X1 CHARGE. FOR EACH X1, PST GOES FROM
C      0.0 TO PSIMAX. THEN X1=X1+DX1 AND THE
C      PROCESS CONTINUES UNTIL X1=XMNAX.
C
C
C      400 L=0
C          E =1
C          S =0
C          Y =0.
C          YE=0.0
C          AY=0.0
C          AT=0.0
C          SU_1=0.0
C          SU_2=0.0
C          SC1=0
C          T1=0.
C          T2=0.
C          SH1=0.
C          CH2=0.
C          PSBAM=SHAM
C          I =((1.+5*PSBAM)/E)-BAR
C          L=I-N1
C          IF (L < 0) GOTO 31
C          CALL RAIK1 (PST,X1,PS1,I,FOUT,PSBAM,-LYC,AU,HI)
C          I =I+
C          IF (I > N1) GOTO 15
C          I =1
C          15 CALL RAIK1 (AU,NSC,EPSBAM,PSBAM,F,Y,YP,FX,AY,AYP,SUM1,SUM2,I)
C          CALL RAIK1 (PST,X1,PS1,I,FOUT,PSBAM,PSHAR,PLYC,AU,HI)
C          I=I+N1+1
C          IF (ANS(F).LT.1.UF-36) GOTO 21
C          20 Y=AY
C          FX=0.
C          21 IF (I .EQ. 4) GOTO 15,16
C          16 GO TO 15
C          C ANY DESIRED CODING
C          11 GO TO 15
C          Z1=FX1*(Z.*X1)
C          FX=Z1*Y
C          IF (FX .GT. 1.) GOTO 6
C          6 FX=1.
C          7 US1A=Z1,-FX*(1.0-0.07/E)
C          I=US1A+E
C          US1B=1.0*(R51)*(US1A-E)/(1.0-0.07/E)+1.0*(R51)*(1.0-US1A)/(1.0-0.07/E)
C          US1B=US1B/(US1A+E)
C
C
C          SR1L ((1.,432)*PS1,X1,FX,PSHAR,0,C,0,L,KD,Y
C          432 + 0.01A ((//,6H PST E17.9,4X,5HX) E17.9,3X,
C          134H A1D VALUES AFTER FIRST INTEGRATION//,
C          16H FX E17.9,4X,6H PSHAR E17.9,4X,
C          26H E17.9,4X,6H UNI-HOK E17.9,4X,
C          66H Y E17.9//)
C
C
C          END OF RETURN FOR T EVALUATION
C
C
C      500 CONTINUE
C
C      RETURN FOR FREE VALUE EACH EVALUATION

```

```

C
      DO 10 I=1,20
      TAU(I)=0.0
10    DO 11 J=1,8
      DO 12 I=1,4,10
      /14=0.0
      /15=0.0
      NW=NWT(I)
      NEX=APZ(I,NW)
      DO 13 I=1,11,KT
      DO 14 I=1,11,KT
/11   TAU(I)=STAH(I,INDEX)
      CALL LINR(I,I,TAU,TAE,KT,2,YI)
      IF (YI) 252,253,253
253   YI=10.0*YI
      DO 15 I=1,254
252   ST=YI
      NST=1
      FNS=FV1
      FFS=AHS(SF)-AHS(FNS)
      YI=10.0*FFS+10.0*FNS
254   DO 16 I=1,KT
      KFL(I)=YI
      DO 17 J=1,NSMAX
      /18=ALFHF(I,J)
      /19=VALHF(I,J)
      /20=LHFHF(I,J)
      DO 20 I=14,15
      /21=Z18-Z19
      KFL(I)=KFL(I)*(82.75*T)**Z15
277   DO 18 I=1,KT
C
C
      EKUL(I),421)(EKUL(I),I=1,NHU)
421   FORMAT (20x,7HEKUL(I)/,(5E22.9))
      EKUL(I),422)(EKUL(I),I=1,NHU)
422   FORMAT (20x,7HEKUL(I)/,(5E22.9))
      EKUL(I),447)
447   FORMAT (//)
C
C      ZERO ARRAY(I,J)
C
D      DO 230 J=1,21
D      DO 230 I=1,22
D      ARA(I,J)=0.0
230   DO 231 I=1,KT
C
C      EQUATIONS 1,...,NNU IN ARAY(I,J)
C      FILE IN ARAY(I, NSMAX2) AND
C      ARAY(I,NSMAX1)
C
      DO 310 I=1,NHU
      /1=0.0
      /2=0.0
      DO 311 J=1,NSMAX
      /2=VALHF(I,J)+Z2
      /1=VALHF(I,J)+Z1
311   DO 312 I=1,KT
      ARA(I,NSMAX2)=Z1-Z2
310   DO 313 I=1,NSMAX
      /1=0.0
      DO 313 J=1,NSMAX

```

```

      SMAX=SMASS(1)
      Z1=Z1+(ALPHPR(I,J)-VALPHR(I,J))*ALPHG(SMU)
313  GO TO 111E
      FKL=L+FKL(I)
      ARHAT(I,NSMAX1)=-/1-ALPHG(FKL,J)
312  GO TO 111E
C
C      IF (D*ARHAY(I,J), I=1,...,NRD AND J=1,...,NSMAX
C
C      DO 314  I=1, NRD
314  DO 314  J=1, NSMAX
      ARHAY(I,J)=.LPHPR(I,J)-VALPHR(I,J)
314  GO TO 111E
C
C      SET-UP ARHAY(I,J), I=NR+1, J=1,...,NSMAX
C
      LPHPR(EF*EF+EF*EF*V1)
      DO 315  I=1, NSMAX
      ARHAY(NR+1,I)=1.0/(CONP*SMASS(I))
315  GO TO 111E
      ARHAT(NR+1,NSMAX1)=CONP*RDZG*ONP
      ARHAT(NR+1,NSMAX2)=-1.
C
C      SET-UP S-R EQUATIONS
C
      DO 347  KNR=1, NR
      K=NR-VK(KNR)
      V14=K+NR+1
347  ARHAT(V14,KNR)=1.0
C
      DO 1260  KNF=1, VRD
      K=NR-VK(KNF)
      DO 1260  I=1, NRD
      DO 1260  J=1, NSMAX
      IF ((AA2(I,J))1113,1113,1112
1113  GO TO 251
1112  IF X=IAA2(I,J)
      II=1
1114  IF ((K-P-KNPF)(II,1DE9))1116,1115,1116
1116  II=II+1
      IF ((I-4)1114,1114,251
1115  Z2=VALPHR(I,KNP)
      Z1=VLPHP(I,1DE9)
      V15=VLT+NR-Z1
      ARHAT(V15,1DE9)=(Z2*SMASS(KNP))/(Z1*SMASS(1DE9))
251  GO TO 111E
250  GO TO 111E
1260  GO TO 111E
C
C      SET RH(I,(KNP+1),...,K(NSMAX)) INTO ARHAY(I,NSMAX+1)
C
      DO 271  I=NR+2,NSMAX1
      J=I-NR+1
      ARHAY(I,NSMAX1)=-01(J)*(0STAR-01)/J/(1.+01/JE)-
      1+1(J)*(1.-0STAR)/(1.+01/JE)
271  GO TO 111E
C
C
      DO 403  I=1, NSMAX1
      ARHAY(I,(404))=(ARHAY(I,J), J=1,NSMAX2)
404  ENDHAT(20,404)12,14H 0F ARHAY(I,J)/(5E22.9)
403  ENDHAT(12,14H)

```

```

      WRITE(10,447)
      TOT=TOT0
      ANEW=0
      CALL MATL2(CNHRM,ANHAR,FUN10N,ARRAY,RELX,SMAX,XHUS,I0L,I1,
     1CNHRM,XHUS,SMAS,LT,IR,REFV,PREF,TREF,XG)
      IF (ANEW)1001,1004
1004 IF (XG)1000,1000,600
      600 DO 93 I=1,1
      600 DO 93 J=1, SMAX
      STAB(I,J)=SSTAB(I,J)
      93 CONTINUE
      600 DO 1002
1001 CONTINUE
C
      WRITE(10,446)(CHAR(I),I=1,SMAX1)
446 FORMAT (/>20X,7HCHAR(I)/,>(5E2.9))
C
C      CHECK THE INTERACTION THROUGH THE ENTHALPY
C
      /1=0.0
      /2=0.0
      UBAR=BAR(SMAX1)
      E=(H1*(USLAR-UJ/RE)+HJ*(1.-USTAR))/(1.-UJ/ RE)
      UBAR=1./E*(USTAR**2*LE**2*2.3901E-8
      BAR(SMAX1)=UNITHC*CVTRM
C
C      FIRST FIND INTEGRATED PART OF SMALL H
C
      /E=1
      /1=0.0
      /2=0.0
      NS=0
      DO 130 N=1,SMAX1
      YP=0.0
      EK=0.0
      AY=0.0
      SUM1=0.0
      SUM2=0.0
      I=0
      R2=0.
      TT=1.0
      IF (I)130,130,131
130  CALL MAIZ(FBUT),SMAS,COH,R,AAUDN,IT,N,AAHUF,SMAX1,CNTIRM)
131  I=1+I
132  DO 111 K=N,V,N
      INV=1
      115 CALL RUMA1(NB0,NSU,EL,TT,F2,Y2,YR,EK,AY,ATR,SCM1,SUM2,INU)
      CALL MAIZ(FBUT),SMAS,COH,R,AAUDN,IT,N,AAHUF,SMAX1,CNTIRM)
      INV=INV+1
      IF (INV=4)115,115,116
116  DO 117
117  CONTINUE
      Y21=Y2
      Y22=Y21
      Y21=Y2
      IT1=IT
      IF (ABS(IT-IT1)<.00000001)161,161,162
162  IF ((IT-IT)650,650,501
C
C      A POINT IN ERRICATION IN Y2
C
      501 Y2=Y2+(IT-IT2)*(IT1-IT2)*(Y21-Y22)

```

```

161 IF(T>=0.0)HAN(4)+=1
      CALL ACAL2(M11,n,AHCL,T0,T4,AACM1,NMAX1)
      IF ((--NSMAX1)*HCL>479,479
      GO TO 478
478 T2=HAN(1)*HCL/NSMAX1+Z2
479 T2=HAN(1)*HCL/NSMAX1+Z2
480 C0,T1,N1,E
      IF(E)
        WHILE ((I>499)CUP,FU1,I2,I1,FU1,T2
499  FORMAT (/,SH 00-Pe17.9,4X,SH Hu-Te17.9,4X,SH T2 E17.9,
     14X,5,11 E17.9/,5F F0,TB17.9,4X,5H N 15/)
150 C0,T1,N1,E
      SH=Z1+Z2
      WHILE ((I>438)SH,NE
433 FORMAT (20H VALUES USED IN THE ENTHALPY,4X,
     15HSH E17.9,4X,5HFF E17.9//)
      C01 IS USED TO DETERMINE CUT-OFF APPROXIMATION
      ABOVE THE H-PRIME VALUE.
      SH2, SH1, T02, T01 AND FF ARE USED IN INTERPOLATION
      IN WHICH GIVES SHFFP.
      IF(AHS(SH2-SH))=.01)35,35,2197
      35 IF (SH<2198,2198,2199
2198 IF (AHS(SH2)-ABS(HP))E21,812,812
2199 IF (AHS(SH2)-ABS(HP))E12,812,821
2197 SH2=-SH1
      SH1=SH
      T02=T01
      T01=
      IF ((T01-S)E40,800,840
840 IF (ABS(HP-SH)-ACUTUL)E50,850,1197
1197 IF (HP)400,400,496
400 IF (SH)E41,841,812
498 IF (SH)E21,841,1199
1199 IF (HP-SH)E22,850,811
641 IF (ABS(HP)-ABS(SH))E01,850,802
801 C01=1
      IF (ABS(HP-SH)-ACUTUL)E20,850,810
810 IF (ABS(HP)-ABS(SH))E11,850,812
811 T=1+0.01*TAT
      GO TO 500
812 T=T02+(AHS(HP-SH2))/(AHS(SH1-SH2))*ABS(C01-T02)
      K01=3
      GO TO 500
802 K01=2
      IF (ABS(HP-SH)-ACUTUL)E20,850,820
820 IF (ABS(HP)-ABS(SH))E21,850,822
821 T=T01+(AHS(HP-SH1))/(AHS(SH2-SH1))*ABS(C02-T01)
      C01=3
      GO TO 500
822 T=T-TELTAT
      GO TO 500
850 C0,T1,N1,E
      CALCULATIONS FOR PRINT-OFF
712=0.0
      RHS1=RH/RH00
      IF ((RHS1)E10,610,610,611
610 RHS1=0.0
      RHS1=(2.+S1)/((CHH/RH00)*(1/U1))
      S1=1
      611 C0,T1,N1,E

```

```

      SFS11=FS12
      SF12=(2.*PSI)/((CEB/R+0J)*(U/U))
      LFY=S12-FS11
      U=U+LFY*(1.+(FSF11+1./2.*DFY+(1./3.-1./2.)*1./2.*1./4.+2*(1./4.)*DFY**3/-AC1(3)+(1./5.-3./2.+11./3.-3./2.)*2*DFY**4/FAU-(4)*(1./6.-2.+35./4.-50./3.+12.)*1.F**5/3/FAU-(5)*(1./1.-15./6.+17.-225./4.+274./3.-60.)*1.LFY**6/FAU(6))
      HSTARH=S12
      HSTAR=HGT(HSTAR)
      CALL LAL (1,1,XSTAR)
      XSTAR=HGT(XSTAR)
      RHE=HSTAR-H
      IXSTAR=0
      *FINAL PRINT STATEMENT*
      ANTHR ((L,300)T,H,SH,H*HSTAR,HSTARH,XSTARH,HHH,A,
      FORMAT (/,,UX,20H)VAL PRNT STATEMENT FOR I=17.9//,
      /H H E17.9,4X,6HSH E17.9,4X,
      6HHP E17.9,4X,6HPP A E17.9//,
      /H HSTARH E17.9,4X,6HXST A SE17.9,4X,
      EHHR E17.9,4X,6HX E17.9//)
      *
      F (-8C(F$)-PSI#4X)-.00000011111111195,1195,2196
      F (+S1-FS1#4A)1196,1195,1195
      S1=S1+FS1
      U DU 1E11,NSMAX
      G(I)=UBAH(I)
      ON TINIE
      K1=(10,95)(AG(I),I=1,NSMAX)
      U DU 400
      F (ABS(X1-X1MAX)-.00000001)1000,1000,2194
      F (X1-X1MAX)1194,1000,1000
      SI=PSI0
      PSI=0
      UME=0.0
      I=X1+LXE
      U DU 400
      N1

```

**3200 FORTRAN DIAGNOSTIC RESULTS - FOR 4000**

Digitized by srujanika@gmail.com

SUBROUTINE NEWTON (CHAR,ARRAY,FUNION,PARRAY,DELX,NSMAX,NRD,EL,  
 1T,XMIN,XMAX,SMASS,CNT,HD,REFV,PREF,TREF,XG)  
 C  
 CBAR(21) ARE APPROXIMATE ANSWER WHICH  
 ARE TO BE USED FOR THE NEWTON METHOD  
 ARRAY(21,22) USED IN EVALUATING THE GIVEN FUNCTIONS  
 FUNCTION(Z1) IS THE VALUE OF FUNCTIONS  
 PARRAY(21,22) IS AN ARRAY OF PARTIALS USED  
 IN NEWTONS METHOD  
 DELX(21) IS DELTA X'S USED TO SET-UP  
 THE NEW CHAR VALUES  
 UNITHD IS USED IN NR(1) EQUATION  
 SMASS(1) IS CONCENTRATION OR MASS OF SPECIES 1  
 XG(1) IS THE INITIAL GUESS FOR CCHAR(1)  
 XGMIN IS THE SMALLEST VALUE THAT  
 XG CAN TAKE IN NEWTON.(INPUT VALUE)  
 COMMON X1U,IN,IO,AEX  
 DIMENSION CCHAR(21),ARRAY(21,22),FUNION(21),  
 1PARRAY(21,22),DELX(21),XX(21),XX1(21)  
 DIMENSION EXPAX(21),SMASS(20),XG(20),DELXI(21)  
 SJ(1)=0.0  
 NSMAX=NSMAX+1  
 NSMAX2=NSMAX+2  
 NRD1=NRD+1  
 NRD2=NRD+2  
 TOLNRW=TOL\*1.001  
 Z1=0.1140  
 Z2=H\*EF\*TREF\*REFV/T  
 DO 222 I=1,NSMAX  
 Z1=XG(1)/SMASS(1)+Z1  
 222 CONTINUE  
 RHOVUE=Z2//1  
 CCHAR(NSMAX1)=RHOVUE  
 LU 115 I=1,NSMAX  
 CCHAR(I)=XG(I)  
 115 CONTINUE  
 C  
 SET XX(1)=CCHAR(1)  
 LU 1 I=1,NSMAX1  
 XX(1)=CCHAR(I)  
 1 CONTINUE  
 LU 116 I=1,NSMAX1  
 IF (XX(1)-XGMIN)>17,117,119  
 119 XX(1)=ALG1(XX(1))  
 GO TO 116  
 117 XX(1)=XG1(XX(1))  
 XX(1)=ALG2(XX(1))  
 116 CONTINUE  
 1000 WRITE(10,100)(XX(I),I=1,NSMAX1)  
 100 FORMAT(10H XX VALUES/, (9E22.9))  
 LU 41 I=1,NSMAX1  
 EXFXX(I)=EXP(XX(I))  
 IF (EXFXX(I)-10.)41,41,902  
 41 CONTINUE  
 C  
 FIND THE VALUE OF FUNION(I), I=1,...,NRD  
 LU 2 I=1,NRD  
 Z1=0.0  
 LU 3 J=1,NSMAX  
 Z1=Z1+ARRAY(I,J)\*XX(J)  
 3 CONTINUE  
 FUNION(I)=Z1+ARRAY(I,NSMAX1)+ARRAY(I,NSMAX2)\*X+(NSMAX1)  
 2 CONTINUE

```

C      FIND THE VALUE OF FUNKN(I), I=NNU+1
    /1=0.0
    DO 4 J=1,NMAX
    Z1=Z1+ARRAY(NNU,J)*EXPXX(J)
    4 CONTINUE
    FUNKN(NNU)=Z1+ARRAY(NNU,NMAX1)-EXP(-XX(NMAX1))
    FIND THE VALUE OF FUNKN(I), I=NNU+2,...,NSMAX+1
    DO 5 I=NNU+2,NSMAX+1
    /1=0.0
    DO 6 J=1,NMAX
    Z1=Z1+ARRAY(I,J)*EXPXX(J)
    6 CONTINUE
    FUNKN(I)=Z1+ARRAY(I,NMAX1)
    5 CONTINUE
C      STORE FUNCTION VALUES IN PARRAY
    DO 35 I=1,NSMAX1
35 PARRAY(1,NSMAX2)=FUNKN(I)
    IF(ABX)26,25,26
25 WRITE (10,27)(FUNKN(I),I=1,NSMAX1)
27 FORMAT (11H FUNCTION I/, (5E22.9))
26 CONTINUE
C      PARTIALS OF FUNCTIONS
    DO 7 I=1,NNU
    DO 8 J=1,NMAX
    PARRAY(I,J)=ARRAY(I,J)
    8 CONTINUE
    PARRAY(I,NSMAX1)=ARRAY(I,NSMAX2)
    7 CONTINUE
    DO 9 J=1,NMAX
    PARRAY(NNU,J)=ARRAY(NNU,J)*EXPXX(J)
    9 CONTINUE
    PARRAY(NNU,NMAX1)=EXP(-XX(NMAX1))
    DO 10 I=NNU,NSMAX1
    DO 11 J=1,NMAX
    PARRAY(I,J)=ARRAY(I,J)*EXPXX(J)
    11 CONTINUE
    PARRAY(1,NSMAX1)=0.0
    10 CONTINUE
    IF (ABX)52,52,54
52 DO 50 I=1,NSMAX1
    Z1=0.
    DO 51 J=1,NSMAX1
    Z1=Z1+ABS(ARRAY(I,J))
    51 CONTINUE
    IF (Z1-1.0)50,50,54
50 CONTINUE
    WRITE (10,55)
55 FORMAT (10H,XMAX IMPLIES C0, ERGENCE//)
    C0,1=1.0
54 CONTINUE
    IF(ABX)31,30,31
30 WRITE (10,32)((PARRAY(I,J),I=1,NSMAX1),J=1,NSMAX2)
32 FORMAT (14H PARIAL ARRAY/, (5E22.9))
31 CONTINUE
    CALL STHW(PARRAY,NSMAX1,DEUX)
    DO 12 I=1,NSMAX1
    X1(I)=XX(I)+DEUX(I)
    12 CONTINUE
    IF(ABX)14,13,14
13 WRITE (10,12)(EX,XX(I),I=1,NMAX1)
    WRITE (10,16)(EX,X1(I),I=1,NSMAX1)
    WRITE (10,21)(XX1(I),I=1,NSMAX1)

```

```

15 FORMAT (10H XX VALUES/, (5E22.9))
16 FORMAT (11H LELA X(1)/, (5E22.9))
21 FORMAT (11H XX1 VALUES/, (5E22.9))
14 GO TO 15
15 GO TO 16, SMAX1
16 EX1(I)=EX-E(MX-X(I))
17 A=S(L+X1(I)-1)+NEW1K,18,20
18 GO TO 19, E
19 20 I=1,NSMAX1
21 BMK(I)=BMK(XX1(I))
22 GO TO 19
23 301 I=1,NSMAX
24 G(I)=BMK(I)
301 GO TO 19
302 RETURN
20 40 60 I=1,NSMAX1
41 EX(I)=YX1(I)
60 GO TO 19
61 I= 1000
902 WRITE(1C,903)
903 FORMAT (14H CASE DIVERGES//)
904 N= 1
905 RETURN
END

```

5200 FORTRAN DIAGNOSIS TO RESULTS - FOR

NO ERRORS

3200 FURNITAN (2.1.0)/\*(HIS) / /

SUBROUTINE FILE (PXI, PPI, U, PPSI, ANS, SPSHAR)

C LUNA N XU  
C \* IS THE PRICE OF EVALUATION  
C ANS IS THE ANSWER DESIRED  
C PPSI, PPI, AND XU ARE NEEDED VALUES  
C FOR THE CALCULATIONS  
C XU=(PPI+SPSHAR\*PPSI)/(2.0\*PXI\*PPI/U)  
C IF (XU-20.)>0,2  
5 =1  
ANS=PXI\*\*2/4.0+1.0  
ANS1=ANS  
10 N=N+1  
A,S=A,S+(XU\*\*N/(2.0\*\*N\*FACT(N)))\*\*2  
ANS2=A,S  
ANS1=A,S  
IF (N<200)4,4,2  
4 IF (ABS(ANS1-ANS2)<.00000001)2,2,3  
3 GOTO 10  
2 RETURN  
END

3200 FURNITAN SUBROUTINE RESULTS = FUR - F1.

40 ERRORS

3200 FUJIHAN (2.1.0) (RIS)

C SUBROUTINE MAEV1(SX1, SX1, SPS1U, SFOUT, S-SPBAR, SA, SAG, SPOL)  
COMMON XIU  
SFOUT IS THE OUT EVALUATION TO BE INTEGRATED  
C TENSION RA(20)  
CALL F10(SX1,SPS1U,SFS1,A,SPBAR,SPSHAR)  
IF(XIU=20.)2,2,4  
2 /1=(SPBAR+2/(4.0\*SX1))/SPS1U  
IF(Z1=700.)5,2,3  
3 /2=0.  
66 1 7  
5 /2=EXP(-Z1)  
7 Z3=(SFS1\*\*2/(SPS1U\*\*2))/((4.\*SX1)/SPS1U)  
IF(Z3=700.)19,7,8  
8 /4=0.  
66 1 10  
9 /4=EXP(-Z3)  
10 CALL FULVAL(SA,SA,SFOUT,SPS1U)  
SFOUT=Z2\*Z4\*SFOUT+SFSEAR+A\*SPER  
66 1 6  
4 /1=((-(SFS1\*\*2)-(SFSEAR\*\*2\*SPS1U\*\*2)+2.\*SFSE1\*\*2\*SPS1U  
1\*SFSEAR)/(4.\*SX1\*SFS1U))  
IF(Z1)11,12,12  
11 IF(Z1+700.)13,13,12  
12 /2=EXP(Z1)  
66 1 14  
13 /2=0.  
14 CALL FULVAL(SA,SA,SFOUT,SPS1U)  
EM=SINH((2.\*3.14159265\*SPS1U\*SFSEAR)/(2.\*SX1))  
SFOUT=(Z2\*SPOL\*SFSEAR)/EM  
A RETURN  
END

3200 FUJIHAN DIAGNOSTICS - RESULTS - FOR - MAF..1

NO ERRORS

3200 FURHAN (2.1.0)/\*(RIS)

C SUBROUTINE POLVAL(AAL,A,YY,PFSBAR)  
C AAL IS THE CONSTANT VALUE OF THE POLYNOMIAL  
C A IS THE A(1)...A(20) COEFFICIENTS  
C YY IS THE VALUE OF THE POLYNOMIAL AT PFSBAR  
C LINE SIGN A(20)  
C YY=AAL+(((((A(20)\*PFSBAR+A(19))\*PFSBAR+  
C 1A(18)\*PFSBAR+A(17))\*PFSBAR+A(16))\*PFSBAR+A(15))  
C 2\*PFSBAR+A(14))\*PFSBAR+A(13))\*PFSBAR+A(12))\*  
C 3\*PFSBAR+A(11))\*PFSBAR+A(10))\*PFSBAR+A(9))\*  
C 4\*PFSBAR+A(8))\*PFSBAR+A(7))\*PFSBAR+A(6))\*  
C 5\*PFSBAR+A(5))\*PFSBAR+A(4))\*PFSBAR+A(3))\*  
C 6\*PFSBAR+A(2))\*PFSBAR+A(1))\*PFSBAR)  
C RETURN  
C END

3200 FURHAN V DIAGNOS IC RESULTS - FUR - POLVAL

NO ERRORS

3200 FURIRAN (C,1,0)Z(RTS)

SUPER-DIRECT AAAL1(C,0,0,-HR,AAACG,1)\*+AAAHT(1,1)\*+AAAHUT(1,1)\*+AX1)  
31 C1 TENSION AAAL1((6,20,2),AAAHUT(6,2))  
1F ((1-1000,)1,2,2  
2 IF ((-1\$AX1)11,22,22  
11 C0 P=(AAACG,(1,N,1)\*+AAACG,(2,N,1)\*T)+AAACG,(3,N,1)\*T+\*\*2+  
1AAACG,(4,N,1)\*T+\*\*3+AAACG,(5,N,1)\*T+\*\*4)\*HR  
50 T 33  
22 C0 P=(AAAH-T(1,1)\*+AAAH-T(2,1)\*T+AAAHUT(3,1)\*T+\*\*2+  
1AAAH-T(4,1)\*T+\*\*3+AAAHUT(5,1)\*T+\*\*4)\*HR  
33 C0 T LINE  
RETURN  
1 IF ((-1\$AX1)31,32,32  
31 C0 P=(AAACG,(1,N,2)\*+AAACG,(2,N,2)\*T)+AAACG,(3,N,2)\*T+\*\*2+  
1AAACG,(4,N,2)\*T+\*\*3+AAACG,(5,N,2)\*T+\*\*4)\*HR  
50 T 33  
32 C0 P=(AAAH-T(1,2)\*+AAAH-T(2,2)\*T+AAAHUT(3,2)\*T+\*\*2+  
1AAAH-T(4,2)\*T+\*\*3+AAAH-T(5,2)\*T+\*\*4)\*HR  
33 C0 T LINE  
RETURN  
50 T

3200 FURIRAN DIAGNOSTIC RESULTS - FOR AAAL1

NO ERRORS

```

      3200  FORTRAN (2.1.0)/(HTS)
      SUBR QLIRE MAIR2(EL,LT,SMASS,CO,N,AACON,T,TT,AAHUT,NSMAX1,
     1CUT1,MM)
      L1=EL+LT-1,SMASS(1,1),AACON(1,1),AAHUT(1,1)
      CALL ALAL1(CUT1,N,AACON,TT,N,AAHUT,NSMAX1)
      IF (CUT1>SMAX1)1,2,2
1      ENDIF
2      ENDIF
3      CONTINUE
4      RETURN
      END

```

3200 FUKUOKA DIAGNOSTIC RESULTS - FOR 1981-82

## NO ERRORS

## 3200 FORTRAN (2.1.0)/(HTS)

```

SUBROUTINE ACAL2(HFUT,HR,AAAHOT,TI,N,AAACON,NSMAX)
COMMON AAAHOT(6,2),AAACON(6,20,2)
IF (II-1000,1,2,2
1 K=2
2 DO II=4
3 I=1
4 IF (N-NSMAX1)5,6,6
5 HFUT=(AAACUN(1,N,K)+AAACUN(2,N,K)*TI/2.+AAACUN(3,N,K)*TI*TI/3.+.
1 AAACUN(4,N,K)*TI**3/4.+AAACUN(5,N,K)*TI**4/5.+.
2 AAACUN(6,N,K)/TI)*RR+IT
6 DO II=7
7 HFUT=(AAAHOT(1,K)+AAAHOT(2,K)*TI/2.+AAAHOT(3,K)*TI*TI/3.+.
1 AAAHOT(4,K)*TI**3/4.+AAAHOT(5,K)*TI**4/5.+.
2 AAAHOT(6,K)/TI)*RR+IT
7 CONTINUE
RETURN
END

```

## 3200 FORTRAN DIAGNOSTIC RESULTS - FOR ACAL2

NO ERRORS

3200 FORTRAN (2.1.0)/(RTS)

SUBROUTINE CAL(XI,ANSWER)  
ANSWER = 2.\*XI/.00075  
RETURN  
END

3200 FORTRAN DIAGNOSTIC RESULTS - FOR CAL

NO ERRORS

3200 FORTRAN (2.1.0)/(RTS) / /

```
C      FUNCTION FACT(I)
C      DETERMINES N-FACTORIAL FOR NN
C      IF(I-1)1,1,2
1   FACT=1
      RETURN
2   FACT=1
      DO 3 J=2,I
      FJ=J
3   FACT=FACT*FJ
      RETURN
      END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR FACT

NO ERRORS

## 3200 FORTRAN (2.1.0)/(RIS) / /

```

C   SUBROUTINE RUNKUT(ARG, SO, H, X, F, Y, YP, FK, AY, AYP, SUM1, SUM2, IJDT)    IN000050
C   RUNKUT IS A RUNGE KUTTA INTEGRATION
C   DIMENSION F(100), Y(100), YP(100), AY(100), AYP(100), SUM1(100)      IN000040
C   DIMENSION SUM2(100), FK(100)                                              IN000050
C   THIS INTEGRATES NEQ EQUATIONS, NSO OF WHICH ARE SECOND ORDER.          IN000060
C   H IS THE STEP SIZE, X THE INDEPENDENT VARIABLE. THE DEPENDENT           IN000070
C   VARIABLES ARE STORED IN Y AND YP, SECOND ORDER EQUATIONS FIRST.        IN000080
C   F IS THE VALUE OF THE FUNCTION (FIRST OR SECOND DERIVATIVE).          IN000090
C   AUXILIARY STORAGE MUST BE FURNISHED, FK(NEN), AY(NEQ), AYP(NEQ),       IN000100
C   SUM1(NSO), SUM2(NEQ). SUBROUTINE IS CALLED FOUR TIMES WITH VALUES       IN000110
C   OF X, Y, YP, F. SUBROUTINE FURNISHES VALUES OF X, Y, YP. INDI IS        IN000120
C   COUNTER (1,2,3 OR 4). SUBROUTINE STORES NO ESSENTIAL INFORMATION        IN000130
C   INTERNALLY
C   IF(IJDT-2) 1,2,5
1  X=X+H/2.
AL1=H/2.
AL2=0.
AL3=.5
AL4=1.
GO TO 1,I=1,NEQ
AY(1)=Y(1)
AYP(1)=YP(1)
SUM1(1)=0.
6  SUM2(1)=0.
GO TO 7
2  AL1=H/2.
AL2=H/4.
AL3=.5
AL4=2.
GO TO 7
5  IF (IJDT-3) 3,3,4
3  X=X+H/2.
AL1=H
AL2=H/2.
AL3=1.
AL4=2.
GO TO 7
7  DO 8 I=1,NEQ
     IF (I-NSO) 9,9,10
9  Y(1)=AY(1)+AL1*AYP(1)+AL2*FK(1)
     FK(1)=H*F(1)
     YP(1)=AYP(1)+AL3*FK(1)
     SUM1(1)=SUM1(1)+FK(1)
     GO TO 11
10  FK(1)=H*F(1)
     Y(1)=AY(1)+AL3*FK(1)
11  SUM2(1)=SUM2(1)+AL4*FK(1)
8  GO TO 12
RETURN
4  DO 12 I=1,NEQ
     FK(1)=H*F(1)
     SUM2(1)=(SUM2(1)+FK(1))/6.
     IF (I-NSO) 13,13,14
13  Y(1)=AY(1)+H*AYP(1)+H* SUM1(1)/6.
     YP(1)=AYP(1)+SUM2(1)
     GO TO 12
14  Y(1)=AY(1)+SUM2(1)
12  CONTINUE
RETURN
END

```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR RUNKUT

NO ERRORS

```

C SUBROUTINE CLINT(ARG,X,Y,NPT,IP,YI)
C CARRIAGE INTERPOLATION, 2,3,4,5 POINT
C IP IS DEGREE OF INTERPOLATION
C NPT IS NUMBER OF PLINTS GIVEN IN TABLE
C IP=1:0 X(2),Y(2)
C IP=2
C I=1
20 IF(I-X(1))40,25,30
25 Y=Y(1)
30 G0 10 400
35 I=I+1
40 IF(I-NPT)20,20,35
45 I=IP
50 G0 10 (40,55,60,60,60),IP
C 2 POINT INTERPOLATION
55 I=I-1156,56,57
56 I=I+1
57 G0 10 400
C 3 POINT INTERPOLATION
60 U01=(I-X(1))/(X(1)-X(1))
61 U02=(I-X(1-1))/(X(1)-X(1-1))
62 Y1=U01*Y(1-1)+U02*Y(1)
63 G0 10 400
64 I=I+2
65 I=I+1
66 G0 10 60
110 IF(I+1-NPT)145,145,140
140 I=-1
141 G0 10 110
142 G0 10 (145,145,100,200,300),IP
C 3 POINT INTERPOLATION
100 U1=X(1-1)
U2=X(1)
U3=X(1+1)
143 U01=((I-U2)*(I-U3))/((U1-U2)*(U1-U3))
144 U02=((I-U1)*(I-U3))/((U2-U1)*(U2-U3))
145 U03=((I-U1)*(I-U2))/((U3-U1)*(U3-U2))
Y1=U01*Y(1-1)+U02*Y(1)+U03*Y(1+1)
146 G0 10 400
C 4 POINT INTERPOLATION
200 U1=X(1-2)
U2=X(1-1)
U3=X(1)
U4=X(1+1)
U1=U1-U2
U2=U1-U3
U3=U1-U4
U4=U2-U3
U5=U2-U4
U6=U3-U4
U7=U1-U2
U8=U1-U3
U9=U1-U4
U10=U1-U1
U11=(U1*(U2+U3))/(U1+U2+U3)
U12=-(U4*(U2+U3))/(U1+U4+U5)
U13=(U4*(U1+U3))/(U2+U4+U6)
U14=-(U4*(U1+U2))/(U3+U4+U6)
Y1=U11*Y(1-2)+U12*Y(1-1)+U13*Y(1)+U14*Y(1+1)
147 G0 10 400
C 5 POINT INTERPOLATION

```

300  
   $\begin{aligned} &1=x(1-3) \\ &2=x(1-2) \\ &3=x(1-1) \\ &4=x(1) \\ &5=x(1+1) \\ &6=x(1+2) \\ &7=x(1+3) \\ &8=x(1+4) \\ &9=x(1+5) \\ &10=x(1+6) \\ &11=x(1+7) \\ &12=x(1+8) \\ &13=x(1+9) \\ &14=x(1+10) \\ &15=x(1+11) \end{aligned}$   
   $\begin{aligned} &U1=(11+2+3+14)/(D1+D2+D3+D4) \\ &U2=-(15+12+13+14)/(E1+D5+D6+D7) \\ &U3=(15+11+13+14)/(D2+D5+D8+D9) \\ &U4=-(15+11+12+14)/(E3+D6+D8+D10) \\ &U5=(15+11+12+13)/(D4+D7+D9+D10) \\ &Y1=U1*Y(1-3)+U2*Y(1-2)+U3*Y(1-1)+ \\ &U4*Y(1)+U5*Y(1+1) \end{aligned}$

400 RETURN  
END

3200 FORTRAN DIAGNOSTIC RESULTS - FOR LINT

NO ERRORS

## 3200 FORTRAN (2.1.0)/HTS

```

SUBROUTINE SIMEQ(A,N,C)
FILE=STDIN,A(21,22),B(21,22),C(21)
TOL=5,UE=0B
10 A=N+1
20 I=2
30 J=1
40 LU 10 I=1,N
50 LU 10 J=1,KA
C MOVE MATRIX TO WORKING AREA
60 H(I,J)=A(I,J)
70 IF(AH>B(K3,K3)*Y)-10L)30,100,100
80 IF(K2=7)40,40,300
90 LU 50 I=1,KA
100 H(K3,1)
110 H(K3,1)*H(K2,1)
120 H(K2,1)*H
130 CONTINUE
140 I=2*K3
150 J=K3+1
160 LU 120 I=20,120,200
170 LU 130 I=K3,KA
180 H(K1,1)=H(K1,1)/R(K1,K1)
190 CONTINUE
200 LU 140 I=K3,N
210 LU 140 I=K3,N
220 H(K3)=H(K3,KA)-D
230 K3=K3-1
240 LU 220 I=K1,N
250 =D+H(K3,1)*C(I)
260 C(K3)=H(K3,KA)-D
270 K3=K3-1
280 LU 240 I=350,350,240
290 TOL=TOL/10.
300 IF(TOL<1.0E-30)320,310,310
310 K2=K3+1
320 LU 20
330 RETURN
340 END

```

## 3200 FORTRAN DIAGNOSTIC RESULTS - FOR SIMEQ

NO ERRORS  
LOAD,56  
RUN,6

# LISTING OF THREE-STREAM-MIXING COMPUTER PROGRAM

3200 FORTRAN (2,1.0)/BRTS

```

PROGRAM JETMIX
COMMON XIO,IN,IO,ABX
C A0=CONSTANT COEFFICIENT OF THE POLYNOMIAL
C FLYC=COEFFICIENT OF POLYNOMIAL(PSIBAH)
C AACON(I,J,K)=CONSTANTS USED IN FINDING HOT AND COLD
C AACON(I,J,K)=LOWER TEMPERATURE RANGE FOR K=2 AND
C AACON(I,J,K)=UPPER TEMPERATURE RANGE FOR K=1 AND
C FOR I=1,...,6 AND J=INDEX FOR SPECIE K.
C AAHOT(I,J) WHERE J=1 OR 2 WITH J=2 IS FOR
C THE LOWER TEMPERATURE RANGE FOR NITROGEN.
C TTAB=TABLE OF T
C KTEMU, OF VALUES IN EACH SPECIE TABLE
C STAB(I,J) ARE TABLES OF SPECIE VALUES
C CORRESPONDING TO SPECIE J WHERE I=1,...,KI
C T IS VALUE OF INPUT FOR TABLE LOOK-UP
C NALPH, NALPHP, NAA, NAA2, NSMAX, NP AND NNP ARE USED
C TO SETUP INPUT EQUATIONS
C ET=INCREMENT FOR INTEGRATION
C DELTAT=INCREMENT FOR T ITERATION
C C1(I) AND G1(I) FOR EQUATION I=NRU+1 AND CNITRO
C CBAR(I) IS SOLUTION OF NON-LINEAR SYSTEM OF EQUATIONS
C SMASS(I) IS CONCENTRATION OR MASS OF SPECIES I
C CNITRO IS USED IN KRC+I EQUATION
C CNITROM=MASS OF NITROGEN
C KNEW = 1 CAUSES NEW CASE TO BE READ.
C KNEW IS SET IN NEWTON.
C NXG = 1 IMPLIES THAT NO XG GUESSES ARE AFTER CASE.
C NXG = 0 IMPLIES THAT XG GUESSES ARE AFTER CASE.
C XGMIN IS THE SMALLEST VALUE THAT
C XG CAN TAKE IN NEWTON.(INPUT VALUE)
C
DIMENSION TTAB(20),STAB(20,20)
DIMENSION KSPEC(4,20)
DIMENSION A0C(40),AU(40),KELEM(20),AACON(6,20,2),AAHOT(6,2)
DIMENSION NAA(10,20),NAA2(10,20),NP(15),NNP(10)
DIMENSION NALPH(10,20),NALPHP(10,20)
DIMENSION FKCL(10),FKPL(10),TAB(20),ARRAY(21,22)
DIMENSION XG(20),CBAR(21)
DIMENSION SMASS(20),C1(15),G1(15),PP(6),PY(6)
DIMENSION A0V(40),A0E1(40),A0E2(40),A0E3(40),A0E4(40),A0EE(40)
DIMENSION FUNINT(21),DELTAT(21),PARRAY(21,22)
DIMENSION SSTAB(20,20)
EQUIVALENCE T#DOT,F1,(FDO,T,F2)

C
C READ INPUT
C
C
IN=60
10=61
ABX=0
1000 READ (IN,2)KT,NSMAX,NR0,NXG
I1=1
II0=10
NR01=NR0+1
NR02=NR0+2
NSMAX1=NSMAX+1
NSMAX2=NSMAX+2
NRS=NSMAX1-NR01
NRS1=NRS+1
NFA=0

```

```
HEAD (IN,3)(D1(I),I=1,NRS1)
READ (IN,3)(G1(I),I=1,NHS1)
HEAD (IN,3)(SMASS(I),I=1,NSMAX)
READ (IN,2)((NALPHF(I,J),I=1,NR0J,J=1,NSMAX)
HEAD (IN,2)((NLPHPP(I,J),I=1,NR0J,J=1,NSMAX)
HEAD (IN,3)PSI,DPST,EPSSBAR,XI,DXI,PSIMAX,XIMAX,TREF,PREF,REFV,
1FE,HJ,ACCTOL,PSIJ,T,CELTAT,DT,TOL,PSBAR,UJ,UE,USTARJ,RHOUJ,RJ,
2CNITRM
```

```
HEAD (IN,3)((AACON(I,J,1),I=1,6),J=1,NSMAX)
HEAD (IN,3)((AACON(I,J,2),I=1,6),J=1,NSMAX)
HEAD (IN,3)((AAHOT(I,J),I=1,6),J=1,2)
```

```
HEAD (IN,3)AUC
```

```
HEAD (IN,3)AUV
```

```
READ (IN,3)AUE1
```

```
READ (IN,3)AUE2
```

```
READ (IN,3)AUE3
```

```
HEAD (IN,3)AUE4
```

```
READ (IN,3)AUEE
```

```
READ (IN,3)(TTAB(K),K=1,KT)
```

```
HEAD (IN,3)((STAB(I,L),I=1,KT),J=1,NSMAX)
```

```
1002 READ (IN,3)(XG(I),I=1,NSMAX)
```

```
READ (IN,3)XGMIN
```

```
HEAD (IN,3)SUM
```

```
2 FORMAT (12I5)
```

```
3 FORMAT (6F12.5)
```

```
4 FORMAT (6E20.8)
```

```
C SET UP NAA AND NAA2 ARRAYS FROM NALPHF AND NLPHPP
```

```
C
```

```
DO 470 I=1,NRU
```

```
DO 470 J=1,NSMAX
```

```
; NAA(I,J)=0
```

```
NAA2(I,J)=0
```

```
470 CONTINUE
```

```
DO 450 I=1,NRU
```

```
DO 451 J=1,NSMAX
```

```
IF (NALPHF(I,J)) 452,453,452
```

```
452 NAA(I,J)=J
```

```
453 IF (NLPHPP(I,J)) 454,451,454
```

```
454 NAA2(I,J)=J
```

```
451 CONTINUE
```

```
450 CONTINUE
```

```
C
```

```
C SET UP NP AND NNP ARRAYS FROM NALPHF AND NLPHPP
```

```
C
```

```
K=1
```

```
J=0
```

```
456 J=J+1
```

```
DO 455 I=1,NRU
```

```
IF (NALPHF(I,J)) 455,455,458
```

```
458 NP(K)=J
```

```
K=K+1
```

```
IF (K=NRS) 456,456,459
```

```
455 CONTINUE
```

```
IF (K=NRS) 456,456,459
```

```
459 CONTINUE
```

```
K=1
```

```
J=0
```

```
466 J=J+1
```

```
DO 465 I=1,NRU
```

```

O   IF (NLPHPP(1,J)) 465,465,468
O   468 KNP(K)=J
O   k=k+1
O   IF (K-NR0) 466,466,469
O   465 CONTINUE
O   IF (K-NR0) 466,466,469
O   469 CONTINUE
C   C PRINT INPUT
C
O   WRITE (10,91)KT,NSPAX,NR0
O   91 FORMAT (1H1,/,28H JET MIXING JOB NO. 560180//,
O   131H NUMBER OF ELEMENTS IN TABLES =I3,4X,
O   220HNUMBER OF UNKNOWNS =I3,4X,
O   327HNUMBER OF INPUT EQUATIONS =I3//)
O   WRITE (10,92)(D1(I),I=11,NRS1)
O   92 FORMAT (6H D1(I)/,(5E22,9))
O   WRITE (10,86)(G1(I),I=11,NRS1)
O   86 FORMAT (6H G1(I)/,(5E22,9))
O   WRITE (10,94)(SMASS(I),I=1,NSMAX)
O   94 FORMAT (18H MASS OF SPECIES I/,,(5E22.9))
O   WRITE (10,95)(XG(I),I=11,NSMAX)
O   95 FORMAT (13H CBAR GUESSES/,,(5E22.9))
O   WRITE (10,352)XGMIN
O   352 FORMAT (7H XG MINIMUM VALUE,5X,E17.9//)
O   WRITE (10,96)((NALPHF(I,J),I=11,NR0),J=1,NSMAX)
O   96 FORMAT (7H NALPHF/,,(10I10))
O   WRITE (10,97)((NLPHPF(I,J),I=11,NR0),J=1,NSMAX)
O   97 FORMAT (7H NLPHPF/,,(10I10))
O   WRITE (10,98)((NAA(I,J),I=11,NR0),J=1,NSMAX)
O   98 FORMAT (9H NAA(I,J)/,(10I10))
O   WRITE (10,99)((NAA2(I,J),I=11,NR0),J=1,NSMAX)
O   99 FORMAT (10H NAA2(I,J)/,(10I10))
O   WRITE (10,80)(NP(I),I=11,NRS)
O   80 FORMAT (6H NP(I)/,(10I10))
O   WRITE (10,81)(NNP(I),I=11,NR0)
O   81 FORMAT (7H NNPTIT//,(10I10))
O   WRITE (10,82)PSI,DPSI,DPSBAR,XI,DXI,PSIMAX,XIMAX,
O   1TREF,PREF,REFV,HE,HJ,ACCTOL,PSIJ,T,DELTAT,DT,TOL,
O   2PSAR,UJ,UE,USTARJ,RHOJ,RJ,CNITRM
O   82 FORMAT (7H PSI E17.9,4X,5H DPSI E17.9,2X,7H DPSIBARE E17.9,
O   14X,5HXI E17.9,4X,5HDXI E17.9,/,6H PSIMXE17.9,4X,
O   25HXIMAXE17.9,4X,5HTREF E17.9,4X,5HPREF E17.9,4X,
O   35HREFV E17.9/,6H HE E17.9,4X,5HHJ E17.9,3X,
O   46MACUTULE17.9,4X,5HPSIJ E17.9,4X,5HT E17.9/,
O   56H DELATE17.9,4X,5HDT E17.9,4X,5HJUL E17.9,
O   13X,6HPSIBARE17.9,4X,5HUJ E17.9/,6H JE E17.9,
O   73X,6HUSTARJE17.9,4X,5HKKHJ E17.9,4X,5HRJ E17.9,3X,
O   86HCUNITRM E17.9//)
C
O   EO 477 K=1,2
O   EO 431 J=1,NSMAX
O   WRITE (10,83)J,(AACON(I,J,K),I=1,6)
O   83 FORMAT (20X,15HAACON(I,J,K) J=12/,,(5E22.9))
O   431 CONTINUE
O   IF (K-2) 473,472,472
O   473 WRITE (10,476)
O   476 FORMAT (/,19X,35H AACON ARRAY FOR LOWER TEMPERATURES/)
O   477 CONTINUE
O   472 CONTINUE
O   WRITE (10,489)TARMCT(I,1),I=1,6
O   489 FORMAT (/,20X,5HAAHOT/,,(5E22.9))

```

```

      WRITE (IO,475)(AAHCT(I,2),I=1,6)
475 FORMAT (/,20X,28HAAHCT FOR LOWER TEMPERATURES/, (5E22.9))
      WRITE (IO,84) AUC
      WRITE (IO,84) AUV
      WRITE (IO,84) AUE1
      WRITE (IO,84) AUE2
      WRITE (IO,84) AUE3
      WRITE (IO,84) AUE4
      WRITE (IO,84) AUEE
84 FORMAT (//,19X,13H AC,A1,...,AN/, (5E22.9))

C   PRINT TABLES
C
      WRITE (IO,193)(TAB(K),K=1,KT)
193 FORMAT (//,20X,7HT TABLE//, (5E22.9))
      WRITE (IO,444)
444 FORMAT (//,20X,32HSPECIE TABLES WITH X=LOG(NUMBER)//)
      DO 200 I=1,NSMAX
      WRITE (IO,194) I
194 FORMAT (20X,9HSPECIES = [3,6H TABLE])
      WRITE (IO,195)(STAB(K,I),K=1,KT)
195 FORMAT (5E22.9)
200 CONTINUE

C   SETTING KELEM(I)=1
C
      DO 5 K=1,20
      KELEM(K)=K
5 CONTINUE

C   CONVERT S1AH,...,S20TAB TABLES FROM X=LOG(Y) TO Y VALUES
IF (NFA)100,350,100
100 IF (NXG)350,350,351
350 NFA=1
      DO 255 I=1,KT
      LU 255 J=1,NSMAX
255 SSTAB(I,J)=STAB(I,J)

C
      351 DO 260 J=1,NSMAX
      DO 260 I=1,KT
      IF (STAB(I,J))493,491,491
491 STAB(I,J)=10.0**STAB(I,J)
      GO TO 490
493 ST=STAB(I,J)
NST=ST
FNST=NST
FFST=ABS(ST)-ABS(FNST)
STAB(I,J)=10.0**FFST+10.0**FNST
490 CONTINUE
260 CONTINUE

C
      WRITE (IO,445)
445 FORMAT (//,20X,27HSPECIE TABLES WITH X=NUMBER//)

C
      DO 401 J=1,NSMAX
      WRITE (IO,402) J,(STAB(I,J),I=1,KT)
402 FORMAT (20X,13HSPECIE TABLE 12/, (5E22.9))
401 CONTINUE
      WRITE (IO,219)
219 FORMAT (//,20X,24HKSPEC(I,J) WHERE J=INDEX//)
      DO 88 J=1,NSMAX
      WRITE (IO,89)(KSPEC(I,J),I=1,4)

```

```

88 CONTINUE
89 FORMAT (3X,4I5)
DO 34 I=1,40
AO(I)=0.
34 CONTINUE
C
C   FREDECK VALUES
C
KPSI=0
TOL=TOL
FSHARU=PSBAR
FSIO=PSI
X10=X1
T0=300.
R=1.987
C
C   900 IS THE RETURN POINT FOR PSI AND
C   XI CHANGE. FOR EACH XI, PSI GOES FROM
C   FSIO TO PSIMAX. THEN XI=XI+DXI AND THE
C   PROCESS CONTINUES UNTIL XI=XIMAX.
C
C
900 MC1=0
TT1=0.
TT2=0.
SH1=0.
SH2=0.
KLL=0
LU 31 IL=1,6
IF(KLL-1)7,8
6 LU 9 I=1,40
AU(I)=AUV(I)
9 CONTINUE
KLL=1
GO TO 30
7 LU 10 I=1,40
AU(I)=AUE1(I)
10 CONTINUE
KLL=2
GO TO 30
8 IF(KLL-3)12,13,14
12 LU 17 I=1,40
AU(I)=AUE2(I)
17 CONTINUE
KLL=3
GO TO 30
13 LU 18 I=1,40
AU(I)=AUE3(I)
18 CONTINUE
KLL=4
GO TO 30
14 IF(KLL-5)19,20,20
19 LU 21 I=1,40
AU(I)=AUE4(I)
21 CONTINUE
KLL=5
GO TO 30
20 LU 22 I=1,40
AU(I)=AUE5(I)
22 CONTINUE
30 NEU=1

```

```

      S=0
      E=0
      SNAME=$NAME
      I=(1+,$5*PSBAR)/EPSBAR
      X=0
      P=0.0
      AY=0.0
      ATY=0.0
      J=1=0.0
      L=240.0
      R(L)=24,70,24
      23 CALL MAJ1(PST,XI,PSIU,PDU,PSHAR,AUC,AU,FUL)
      =1
      24 DO 11 K=1,100
      K=1
      15 CALL HINFL(PSI,Y,PSIU,PSBAR,F,Y,YP,FK,AY,ATY,SUM1,SUM2,I)
      CALL MAJ1(PST,XI,PSIU,PDU,PSHAR,AUC,AU,FUL)
      INC=INH4]
      IF(ABS(F).LT.1.0E-36)32,33
      32 YAY
      P=0.
      33 IF(I.LT.4)15,15,16
      16 CONTINUE
      A*Y=ET*FEM*CPING
      17 CONTINUE
      YAY
      /1=PS1.0/(2.*X1)
      P(1,1)=21*Y
      Y(1,1)=Y
      P(1,1)=10.0/R(1),P(1,1),Y(1,1)
      31 CONTINUE
      USTAR=1.-F(1)*(1.-U0/E)
      E=SSTAR
      USTAR=U(1)*H$1*(1.-F(5))+u1(m$1)*R(5)
      U(1)=U(1)+R(5)*U(1)/E
      C
      C
      C     (J+432)PSI,XI,P(1),USTAR,U,UNITR0,Y(1)
      432 FORMAT (//,64 PSI E17.9,4X,5-X1 E17.9,3X,
      104AD VALUES AFTER FIRST INTEGRATION//,
      11 P(1)E17.9,4X,6E15*AR E17.9,4X,
      20H      E17.9,4X,6E01 RUE17.9,4X,
      6EH Y E17.9V)
      C
      C
      C     PRINT FEM AND FOR T-EVALUATION.
      C
      500 CONTINUE
      C
      C     PRINT THE VALUE EACH EQUATION.
      C
      /10 J=1,20
      -A*(1)=1.0
      /10 T=1,5
      /10 J=1,100
      /14=0.0
      /15=0.0
      /16 N=F(1)
      /17 E=AAZ(1,1,1)
      /10 J=1,100
      /11 I=1,5
      /12 :IAZ(I,1)=SSSTAR(I,INDEX)
      C
      C     N=1.0, U=1.0, T=1.0, E=1.0
      C

```

```

IF (YI)252,253,253
253 YI=10.0*YI
GO TO 254
252 ST=YI
NST=NST
FFST=ABS(ST)-ABS(FNST)
YI=10.0*FFST*10.0*FNST
254 CONTINUE
FKPL(I)=YI
DO 706 J=1,NSMAX
Z16=NALPHP(I,J)
Z13=Z13+Z16
Z15=NLPHP(I,J)
705 Z14=Z14+Z15
Z15=Z13-Z14
FKCL(I)=FKPL(I)*(82.05*T)**Z15
777 CONTINUE
C
C
WRITE (IO,421)(FKCL(I),I=1,NR0)
421 FORMAT (20X,7HKCL(I)/(5E22.9))
WRITE (IO,422)(FKPL(I),I=1,NR0)
422 FORMAT (20X,7HKPL(I)/(5E22.9))
WRITE (IO,447)
447 FORMAT (//)
C
C ZERO ARRAY(I,J)
C
DO 230 J=1,21
TO 230 I=1,22
ARRAY(J,1)=0.0
230 CONTINUE
C
C SET-UP EQUATIONS 1,...,NR0 IN ARRAY(I,J)
C FIRST FILL IN ARRAY(I,NSMAX2) AND
C ARRAY(I,NSMAX1)
C
DO 310 I=1,NR0
Z1=0.0
Z2=0.0
DO 311 J=1,NSMAX
Z2=NALPHP(I,J)+Z2
Z1=NLPHP(I,J)*Z1
311 CONTINUE
ARRAY(I,NSMAX2)=Z1-Z2
310 CONTINUE
DO 312 I=1,NR0
Z1=0.0
DO 313 J=1,NSMAX
SMJ=SMASS(J)
Z1=Z1+(NLPHP(I,J)-NALPHP(I,J))* ALOG(SMJ)
313 CONTINUE
FKCLJ=FKCL(I)
ARRAY(I,NSMAX1)=-Z1-ALOG(FKCLJ)
312 CONTINUE
C
C FIND ARRAY(I,J),I=1,...,NR0 AND J=1,...,NSMAX
C
DO 314 I=1,NR0
DO 314 J=1,NSMAX
ARRAY(I,J)=NLPHP(I,J)-NALPHP(I,J)

```

314 CONTINUE

C C SET-UP ARRAY(I,J), I=NRU+1, J=1,...,NSMAX

C CONP=PREF\*TREF\*REFV/T

LO 315 I=1,NSMAX

ARRAY(NRU1,I)=1./(CONP+SMASS(I))

315 CONTINUE

ARRAY(NRU1,NSMAX1)=CNITRO/CONP

ARRAY(NRU1,NSMAX2)=-1.

C C SET-UP S-H EQUATIONS

C LO 347 NKN=1,NRS

NKNN=NP(NKN)

N14=NKN+NR01

347 ARRAY(N14,NKNN)=1.0

LO 1260 KNCT=1,NRS

KKNP=NP(KNCT)

LO 250 I=1,NRU

LO 251 J=1,NSMAX

IF (NAA2(I,J)) 1113,1113,1112

1113 GO TO 251

1112 IDEX=NAA2(I,J)

I1=1

1114 IF (KKNP-KSPEC(I1,IDEK)) 1116,1115,1116

1116 I1=I1+1

IF (I1-4) 1114,1114,251

1115 Z2=NALPHP(I,KKNP)

Z1=NLPHPP(I,IDEK)

N15=KNCT+NR01

ARRAY(N15,IDEK)=(Z2+SMASS(KKNP))/(Z1+SMASS(IDEK))

251 CONTINUE

250 CONTINUE

1260 CONTINUE

C C STORE K(NRU+1),...,K(SMAX) INTO ARRAY(I,SMAX+1)

LO 271 I=NRU2,NSMAX1

J=1-NRU1

WRITE(10,84)PP(J+1)

ARRAY(I,NSMAX1)=-D1(-)\*(1.-PP(J+1))-G1(J)\*PP(J+1)

271 CONTINUE

LO 403 I=1,NSMAX1

WRITE(10,404)I,(ARRAY(I,J),J=1,NSMAX2)

404 FORMAT(20X,4HROW I2.14H OF ARRAY(I,J)/(,5E22.9))

403 CONTINUE

WRITE(10,447)

TOL=TOLU

+ KNEW = 0

CALL NEWTON(CBAR,ARRAY,FUNION,PAHRY,DELX,NSMAX,NRU,TOL,T,KNEW,

XGMIN,SMASS,CNITRO/REFV,PREF,TREF,XG)

IF (KNEW) 1001,1001,1004

1004 IF (NXG) 1000,1000,600

600 LO 93 I=1,KT

LO 93 J=1,NSMAX

STAB(I,J)=SSTAB(I,J)

93 CONTINUE

GO TO 1002

```

1001 CONTINUE
C      WRITE (IO,446)(CHAR(I),I=1,NSMAX1)
446 FORMAT (/,>20X,7HCHAR(I)/,(5E22.9))
C      CHECK THE ITERATION THROUGH THE ENTHALPY
C
Z1=0.0
Z2=0.0
CBAR=CHAR(NSMAX1)
F=HE*(1.-PP(6))+HJ*PP(6)
HP=H-1./2.*USTAR**2*LE**2*2.3901E-8
CBAR(NSMAX1)=CNITRC+CNITRM
C      FIRST FIND INTEGRATED PART OF SMALL H
C
NEQ=1
Z1=0.0
Z2=0.0
NS0=0
DO 150 N=1,NSMAX1
YP=0.0
FK=0.0
AY=0.0
SUM1=0.0
SUM2=0.0
L=0
Y2=0.
TT=TO
IF (L)130,130,131
130 CALL MAIN2(FDOTT,SMASS,CUP,R,AACON,IT,N,AAHOT,NSMAX1,CNITRM)
131 L=1+L
850 DO 111 KKR=N,N
IND=1
115 CALL RUNKUT(NEQ,NSC,ET,TT,F2,Y2,YP,FK,AY,AYP,SUM1,SUM2,IND)
CALL MAIN2(FDOTT,SMASS,CUP,R,AACON,IT,N,AAHOT,NSMAX1,CNITRM)
IND=IND+1
IF (IND-4)115,115,116
116 CONTINUE
111 CONTINUE
Y22=Y21
Y21=Y2
TT2=TT1
TT1=IT
IF (ABS(T-T1)<.00000001)161,161,162
162 IF (IT-1)650,650,501
C      2 POINT INTERPOLATION ON Y2
C
501 Y2=Y22+((T-TT2)/(TT1-TT2))*(Y21-Y22)
161 Z1=Y2*CBAR(N)+Z1
CALL ACAL2(HOT,R,AAHOT,T,N,AACON,NSMAX1)
IF (N-NSMAX1)480,479,479
480 Z2=CHAR(N)*HOT/SMASS(N)+Z2
GO TO 478
479 Z2=CHAR(N)*HOT/CNITRM+Z2
478 CONTINUE
IT=1
WRITE (IO,499)COP,HOT,Y2,TT,FDOTT,N
499 FORMAT (/,>5H CO-PE17.9,4X,5H HO-TE17.9,4X,5H Y2 E17.9,
14X,5H TT E17.9,5F FDOTT E17.9,4X,5H Y 15/)

150 CONTINUE

```

SH=Z1+Z2  
 WRITE (10,433)SH,HP  
 433 FORMAT (2FH VALUES USED IN THE ENTHALPY,4X,  
 15HSH E17.9,4X,5HFP E17.9///)  
 C KC1 IS USED TO DETERMINE CUT-OFF APPROXIMATE FORM  
 C ABOVE OR BELOW THE H-PRIME VALUE.  
 C SH2,SH1,TC2,TC1 AND FP ARE USED IN INTERPOLATION  
 C ON T WHICH GIVES SH=FP.  
 C  
 IF (ABS(HP-SH)-ACCTOL)850,850,25  
 25 IF (AHS(SH2-SH))-01)35,35,2197  
 35 IF (SH2)2198,2197,2199  
 2198 IF (AHS(SH2)-ABS(HP))E21,812,812  
 2199 IF (AHS(SH2)-ABS(HP))E12,812,821  
 2197 SH2=SH1  
 SH1=SH  
 TC2=TC1  
 TC1=1  
 IF (SH2)830,822,830  
 830 IF (KC1=3)840,850,840  
 840 IF (ABS(HP-SH)-ACCTOL)850,850,1197  
 1197 IF (HP)400,400,498  
 400 IF (SH)841,841,812  
 498 IF (SH)E21,841,1199  
 1199 IF (HP-SH)822,850,811  
 841 IF (ABS(HP)-ABS(SH))E01,850,802  
 801 KC1=1  
 IF (ABS(HP-SH)-ACCTOL)850,850,810  
 810 IF (ABS(HP)-ABS(SH))E11,850,812  
 811 T=T+DELTAT  
 GO TO 500  
 812 T=TC2+(ABS(HP-SH2))/((ABS(SH1-SH2))\*ABS(TC1-TC2))  
 KC1=3  
 GO TO 500  
 802 KC1=2  
 IF (ABS(HP-SH)-ACCTOL)850,850,820  
 820 IF (ABS(HP)-ABS(SH))E21,850,822  
 821 T=TC1+(ABS(HP-SH1))/((ABS(SH2-SH1))\*ABS(TC2-TC1))  
 KC1=3  
 GO TO 500  
 822 T=T-DELTAT  
 GO TO 500  
 850 CONTINUE  
 CALCULATIONS FOR PRINT-OUT  
 Z12=0,0  
 RHSTAR=CBB/RHOJ  
 IF (KPSI)610,610,611  
 610 FSP11=0.0  
 FSP12=(2.\*PSI)/((CBB/RHOJ)\*(U/UJ))  
 KPSI=1  
 611 CONTINUE  
 FSP11=FSP12  
 FSP12=(2.\*PSI)/((CBB/RHOJ)\*(U/UJ))  
 DFY=FSP12-FSP11  
 SUM=SUM+DPSI\*(FSP11+1./2.\*DFY\*(1./3.-1./2.)\*1./2.+  
 1\*DFY\*\*2\*(1./4.)\*DFY\*\*3/FACT(3)+(1./5.-3./2.\*11./3.-3./3)\*  
 2\*DFY\*\*4/FACT(4)+(1./6.-2.\*35./4.-50./3.+12.)\*DFY\*\*5  
 3/FACT(5)+(1./7.-15./8.+17.-225./74.+274./73.-60.)\*  
 4\*DFY\*\*6/FACT(6))  
 RSTARS=SUM  
 RSTAR=SUM/(RSTARS)

```

CALL CAL (XI,XSTARS)
XSTAR=SUPT(XSTARS)
ERR=RSTAR+RJ
X=XSTAR+RJ

C
C   FINAL PRINT STATEMENT
C
      WRITE (10,300)T,H,SH,RHSTAR,RSTARS,XSTARS,RRR,X
300  FORMAT (/,20X,28HFINAL PRINT STATEMENT FOR T=E17.9//,
    17H H E17.9,4X,6HSH E17.9,4X,
    26HMPHIEE17.9,4X,6HSTARSE17.9/,
    37H RSTARSE17.9,4X,6HXSTARSE17.9,4X,
    46HRRR E17.9,4X,6HX E17.9//)

C
C   IF (ABS(PSI-PSIMAX)-.0000001)1195,1195,2196
2196 IF (PSI-PSIMAX)1196,1195,1195
1196 PSI=PSI+DPSI
      LO 50 I=11,NSMAX
      XG(I)=CBAR(I)
50  CONTINUE
      WRITE(10,95)(XG(I),I=11,NSMAX)
      GO TO 900
1195 IF (ABS(XI-XTHMAX)-.0000001)1000,1000,2194
2194 IF (XI-XIMAX)1194,1000,1000
1194 PSI=PS10
      KPSI=0
      SUM=0.0
      XI=XI+DXI
      GO TO 900
END

```

**3200 FORTRAN DIAGNOSTIC RESULTS - FOR JETMIX**

**NO ERRORS**

SUBROUTINE NEWTON (CLEAR,ARRAY,FUNION,MARRAY,DELX,NSMAX,NRO,TCL,  
 1T,KNEW,XGMIN,SMASS,CNITRO,REFV,PREF,IREF,XG)  
 CBAR(21) ARE APPROXIMATE ANSWER WHICH  
 ARE TO BE USED FOR THE NEWTON METHOD  
 ARRAY(21,22) USED IN EVALUATING THE GIVEN FUNCTIONS  
 FUNCTION(21) IS THE VALUE OF FUNCTIONS  
 PARRAY(21,22) IS AN ARRAY OF PARTIALS USED  
 IN NEWTONS METHOD  
 DELX(21) IS DELTA XS USED TO SET-UP  
 THE NEW CBAR VALUES  
 CNITRO IS USED IN KRO+1 EQUATION  
 SMASS(I) IS CONCENTRATION OR MASS OF SPECIES I  
 XG(I) IS THE INITIAL GUESS FOR CBAR(I)  
 XGMIN IS THE SMALLEST VALUE THAT  
 XG CAN TAKE IN NEWTON.(INPUT VALUE)  
 COMMON XIU,IN,IU,ABX  
 DIMENSION CBAR(21),ARRAY(21,22),FUNION(21),  
 1PARRAY(21,22),DELX(21),XX(21),XX1(21)  
 DIMENSION EXPXX(21),SMASS(20),XG(20),DELXI(21)  
 SOL1=0.0  
 NSMAX1=NSMAX+1  
 NSMAX2=NSMAX+2  
 NR01=NRO+1  
 NR02=NRO+2  
 TOLNEW=10L+.001  
 Z1=CNITRO  
 Z2=PREF\*TREF\*REFV/T  
 DO 222 I=1,NSMAX  
 Z1=XG(I)/SMASS(I)+Z1  
 222 CONTINUE  
 RHOVUE=Z2/Z1  
 CBAR(NSMAX1)=RHOVUE  
 DO 115 I=1,NSMAX  
 CBAR(I)=XG(I)  
 115 CONTINUE  
 C SET XX(I)=CBAR(I)  
 DO 1 I=1,NSMAX1  
 XX(I)=CBAR(I)  
 1 CONTINUE  
 DO 116 I=1,NSMAX1  
 IF (YX(I)-XGMIN)117,117,119  
 119 XX(I)=ALOG(XX(I))  
 GO TO 116  
 117 XX(I)=XGMIN  
 XX(I)=ALOG(XX(I))  
 116 CONTINUE  
 1000 WRITE(10,100)(XX(I),I=1,NSMAX1)  
 100 FORMAT(10H XX VALUES/, (5E22.9))  
 DO 41 I=1,NSMAX1  
 EXPXX(I)=EXP(XX(I))  
 IF (EX-XX(I)-10.)41,41,902  
 41 CONTINUE  
 C FIND THE VALUE OF FUNION(I), I=1,...,NRO  
 DO 2 I=1,NRO  
 Z1=0.0  
 DO 3 J=1,NSMAX  
 Z1=Z1+ARRAY(I,J)\*XX(J)  
 3 CONTINUE  
 FUNION(I)=Z1+ARRAY(I,NSMAX1)\*ARRAY(I,NSMAX2)\*XX(NSMAX1)  
 2 CONTINUE

```

C   FIND THE VALUE OF FUNKION(I), I=NRO+1
C   Z1=0.0
C   DO 4 J=1,NSMAX
C     Z1=Z1+ARRAY(NR01,J)*EXPXX(J)
C   4 CONTINUE
C   FUNKION(NR01)=Z1+ARRAY(NR01,NSMAX1)-EXP(-XX(NSMAX1))
C   FIND THE VALUE OF FUNKION(I), I=NRO+2,...,NSMAX+1
C   DO 5 I=NR02,NSMAX1
C     Z1=0.0
C     DO 6 J=1,NSMAX
C       Z1=Z1+ARRAY(I,J)*EXPXX(J)
C   6 CONTINUE
C   FUNKION(I)=Z1+ARRAY(I,NSMAX1)
C   5 CONTINUE
C   STORE FUNCTION VALUES IN PARRAY
C   DO 35 I=1,NSMAX1
C   35 FARRAY(I,NSMAX2)=-FUNKION(I)
C   IF (ABX)26,25,26
C   25 WRITE (10,27)(FUNKION(I),I=1,NSMAX1)
C   27 FORMAT (11H FUNCTION 1/, (5E22.9))
C   28 CONTINUE
C   PARTIALS OF FUNCTIONS
C   DO 7 I=1,NRO
C   DO 8 J=1,NSMAX
C     FARRAY(I,J)=ARRAY(I,J)
C   8 CONTINUE
C     FARRAY(I,NSMAX1)=ARRAY(I,NSMAX2)
C   7 CONTINUE
C   DO 9 J=1,NSMAX
C     FARRAY(NR01,J)=ARRAY(NR01,J)*EXPXX(J)
C   9 CONTINUE
C     FARRAY(NR01,NSMAX1)=EXP(-XX(NSMAX1))
C   DO 10 I=NR02,NSMAX1
C   DO 11 J=1,NSMAX
C     FARRAY(I,J)=ARRAY(I,J)*EXPXX(J)
C   11 CONTINUE
C     FARRAY(I,NSMAX1)=0.0
C   10 CONTINUE
C   IF (SOL1)52,52,54
C   52 DO 50 I=1,NSMAX1
C     Z1=0.
C   DO 51 J=1,NSMAX1
C     Z1=Z1+ABS(PARRAY(I,J))
C   51 CONTINUE
C   IF (Z1-1.0)50,50,54
C   50 CONTINUE
C   WRITE (10,55)
C   55 FORMAT (20X,28HM#MATRIX IMPLIES CONVERGENCE//)
C   SOL1=1.0
C   54 CONTINUE
C   IF (ABX)51,50,51
C   50 WRITE (10,32)((PARRAY(I,J),I=1,NSMAX1),J=1,NSMAX2)
C   32 FORMAT (14H PARTIAL ARRAY/, (5E22.9))
C   31 CONTINUE
C   CALL SIMEQ(PARRAY,NSMAX1,DELX)
C   DO 12 I=1,NSMAX1
C     XX1(I)=XX(I)+DELX(I)
C   12 CONTINUE
C   IF (ABX)14,13,14
C   13 WRITE (10,15)(EXPXX(I),I=1,NSMAX1)
C   WRITE (10,16)(DELX(I),I=1,NSMAX1)
C   WRITE (10,21)(XX1(I),I=1,NSMAX1)

```

```
15 FORMAT (10H XX VALUES/, (5E22.9))
16 FORMAT (11H DELTA X(I)/, (5E22.9))
21 FORMAT(11H XX1 VALUES/, (5E22.9))
14 CONTINUE
   LU 18 I=1,NSMAX1
   LELX1(I)=EXP(DELX(I))
   IF(AHST(DELX1(I))-1.)>TOLNEW)18,18,20
18 CONTINUE
   LU 22 I=1,NSMAX1
   CBAH(I)=EXP(XX1(I))
22 CONTINUE
   LU 301 I=1,NSMAX
   XG(I)=CBAH(I)
301 CONTINUE
   RETURN
20 DO 60 I=1,NSMAX1
   XX(I)=XX1(I)
60 CONTINUE
   GO TO 1000
902 WRITE(10,903)
903 FORMAT (14H CASE-DIVERGES//)
   KNEW = 1
   RETURN
END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR NERTON

NO ERRORS

3200 FORTRAN (2.1.0)/ERTS) / /

SURROUTINE F10 (PXI,FPSIJ,PPSI,ANS,SPSBAR)

COMMON XIO

C X IS THE POINT OF EVALUATION

C ANS IS THE ANSWER DESIRED

C FPSIJ,PPSI AND PXI ARE NEEDED VALUES

C FOR THE CALCULATIONS

XIO=(PPSI\*SPSBAR/FPSIJ)/(2.0\*PXI\*FPSIJ)

IF(XIO<20.0)5,5,2

5 K=1

ANS=XIO\*\*2/4.0+1.0

ANS1=ANS

10 N=N+1

ANS=ANS+(XIO\*\*N/(2.0\*\*N\*FACT(N)))\*\*2

ANS2=ANS1

ANS1=ANS

IF(N>20)4,4,2

4 IF (ABS(ANS1-ANS2)<.00000001)2,2,3

3 GO TO 10

2 RETURN

END

3200 FORTRAN DIAGNOSTIC RESULTS - FOR F10

- NO ERRORS -

## 3200 FORTRAN (2.1.0)/RTS

```

SUBROUTINE MAIN1(SPSI, SXT, SPSIJ, SFDOT, SPSBAR, SAOC, SAU, SPOL)
DIMENSION SAOC(40), SAU(40)
COMMON X10
C SFDO1 IS THE DOT EVALUATION TO BE INTEGRATED
CALL F10(SX1,SPSIJ,SPSJ,ANSWER,SPSBAR)
IF(X10-20.0)2,2,4
2 Z1=(SPSBAR**2/(4.0*SX1))/SPSIJ
IF(Z1-700.)5,5,3
3 Z2=0.
GO TO 7
5 Z2=EXP(-Z1)
7 Z3=(SPSIJ**2/(SPSIJ**2))/((4.*SX1)/SPSIJ)
IF(Z3-700.)9,9,8
8 Z4=0.
GO TO 10
9 Z4=EXP(-Z3)
10 CALL LINT(SPSBAR,SAOC,SAU,40,2,SPOL)
SFDO1=Z2*Z4*SPOL*SPSEAH*ANSWER
GO TO 6
4 Z1=((-(SPSIJ**2)-(SPSEAH**2*SPSIJ**2)+2.*SPSIJ*SPSIJ
1*SPSBAR)/(4.*SX1*SPSIJ))
IF(Z1)11,12,12
11 IF(Z1>700.)13,13,12
12 Z2=EXP(Z1)
GO TO 14
13 Z2=0.
14 CALL LINT(SPSBAR,SAOC,SAU,40,2,SPOL)
EM=SURT((2.*3.14159265*SPSIJ*SPSBAR)/(2.*SX1))
SFDO1=(Z2*SPOL*SPSBAR)/EM
6 RETURN
END

```

## 3200 FORTRAN DIAGNOSTIC RESULTS - FOR MAIN1

NO ERRORS

3200 FORTRAN (2.1.0)/(HTS) / /

C SUBROUTINE POLVAL(AAO,A,YY,PPSBAR)  
C AAO IS THE CONSTANT VALUE OF THE POLYNOMIAL  
C AA IS THE A(1)...A(20) COEFFICIENTS  
C YY IS THE VALUE OF THE POLYNOMIAL AT PPSBAR  
DIMENSION A(20)  
YY=AAO+(((((((((A(20)\*PPSBAR+A(19))\*PPSBAR+  
1A(18))\*PPSBAR+A(17))\*PPSBAR+A(16))\*PPSBAR+A(15))\*  
2PPSBAR+A(14))\*PPSBAR+A(13))\*PPSBAR+A(12))\*  
3PPSBAR+A(11))\*PPSBAR+A(10))\*PPSBAR+A(9))\*  
4PPSBAR+A(8))\*PPSBAR+A(7))\*PPSBAR+A(6))\*  
5PPSBAR+A(5))\*PPSBAR+A(4))\*PPSBAR+A(3))\*  
6PPSBAR+A(2))\*PPSBAR+A(1))\*PPSBAR)  
RETURN  
END

3200 FORTRAN DIAGNOSTIC RESULTS - FOR POLVAL

NO ERRORS

## 3200 FORTRAN (2,1.0)/ERTS

```

SUBROUTINE ACAL1(CCOF,RR,AAACON,TT,N,AAAHT,NSMAX)
DIMENSION AAACON(6,20,2),AAAHT(6,2)
IF (TT-1000.)1,2,2
2 IF (N-NSMAX1)11,22,22
11 CCOF=(AAACON(1,N,1)+AAACON(2,N,1)*TT+AAACON(3,N,1)*TT**2+
1AAACON(4,N,1)*TT**3+AAACON(5,N,1)*TT**4)*RR
GO TO 3
22 CCOF=(AAAHT(1,1)+AAAHT(2,1)*TT+AAAHT(3,1)*TT**2+
1AAAHT(4,1)*TT**3+AAAHT(5,1)*TT**4)*RR
3 CONTINUE
RETURN
1 IF (N-NSMAX1)31,32,32
31 CCOF=(AAACON(1,N,2)+AAACON(2,N,2)*TT+AAACON(3,N,2)*TT**2+
1AAACON(4,N,2)*TT**3+AAACON(5,N,2)*TT**4)*RR
GO TO 33
32 CCOF=(AAAHT(1,2)+AAAHT(2,2)*TT+AAAHT(3,2)*TT**2+
1AAAHT(4,2)*TT**3+AAAHT(5,2)*TT**4)*RR
33 CONTINUE
RETURN
END

```

## 3200 FORTRAN DIAGNOSTIC RESULTS - FOR ACAL1

NO ERRORS

3200 FORTRAN (2.1.0)/MITS

```
SUBROUTINE MAIN2(FDOTT,SMASS,COP,R,AACCN,TT,N,AAHDT,NSMAX1,
1CNITRM)
DIMENSION AACCN(6,20,2),SMASS(20),AAHDT(6,2)
CALL ACAL1(COP,R,AACCN,TT,N,AAHDT,NSMAX1)
IF (N-NSMAX1)1,2,2
1 FDOTT=COP/SMASS(N)
GO TO 3
2 FDOTT=COP/CNITRM
3 CONTINUE
RETURN
END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR MAIN2

NO ERRORS

3200 FORTRAN (2.1.0)/HTS) / /

```
SUBROUTINE ACAL2(HHOT,RR,AAAHOT,TT,N,AAACON,NSMAX1)
LIMENSION AAAHOT(6,2),AAACON(6,20,2)
IF (TT-1000.)1,2,2
1 K=2
GO TO 4
2 K=1
4 IF (N-NSMAX1)5,6,6
5 HHOT=(AAACON(1,N,K)+AAACON(2,N,K)*TT/2.+AAACON(3,N,K)*TT+TT/3.+
1AAACON(4,N,K)*TT**3/4.+AAACON(5,N,K)*TT**4/5.+
2AAACON(6,N,K)/TT)*RR*TT
GO TO 7
6 HHOT=(AAAHOT(1,K)+AAAHOT(2,K)*TT/2.+AAAHOT(3,K)*TT+TT/3.+
1AAAHOT(4,K)*TT**3/4.+AAAHOT(5,K)*TT**4/5.+
2AAAHOT(6,K)/TT)*RR*TT
7 CONTINUE
RETURN
END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR ACAL2

NO ERRORS

3200 FORTRAN (2.1.0)/ERTS) / /

SUBROUTINE CAL(X1,ANSWER)  
ANSWER =2.\*X1/.00075  
RETURN  
END

3200 FORTRAN DIAGNOSTIC RESULTS - FOR CAL

NO ERRORS

3200 FORTRAN (2,1.0)/ERTS

```
C      FUNCTION FACT(I)
C      DETERMINES N-FACTORIAL FOR NN
C      IF(I-1)1,1,2
1     FACT=1.
      RETURN
2     FACT=1.
      DO 3 J=2,I
      FJ=J
3     FACT=FACT*FJ
      RETURN
      END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR FACT

NO ERRORS

## 3200 FORTRAN (2.1.0)/(RTS) / /

```

C SUBROUTINE RUNKUT(NEQ,NSO,H,X,F,Y,YP,FR,AY,AYP,SUM1,SUM2,INDT) IN000030
C RUNKUT=RUNGE KUTTA INTEGRATION IN000020
C DIMENSION F(100),Y(100),YP(100),AY(100),AYP(100),SUM1(100) IN000040
C DIMENSION SUM2(100),FK(100) IN000050
C THIS INTEGRATES NEQ EQUATIONS, NSO OF WHICH ARE SECOND ORDER, IN000060
C H IS THE STEP SIZE,X THE INDEPENDENT VARIABLE. THE DEPENDENT IN000070
C VARIABLES ARE STORED IN Y AND YP,SECOND ORDER EQUATIONS FIRST. IN000080
C F IS THE VALUE OF THE FUNCTION (FIRST OR SECOND DERIVATIVE. IN000090
C AUXILIARY STORAGE MUST BE FURNISHED, FR(NEQ),AY(NEQ),AYP(NEQ), IN000100
C SUM1(NSO),SUM2(NSO). SUBROUTINE IS CALLED FOUR TIMES WITH VALUES IN000110
C OF X,Y,YP,F. SUBROUTINE FURNISHES VALUES OF X,Y,YP, IN000120
C COUNTER (1,2,3 OR 4). SUBROUTINE STORES NO ESSENTIAL INFORMATION IN000130
C INTERNALLY IN000140
C IF (INDT-2) 1,2,5 IN000150
I X=X+H/2. IN000160
AL1=H/2. IN000170
AL2=0. IN000180
AL3=.5 IN000190
AL4=1. IN000200
DO 6 I=1,NEQ IN000210
AY(I)=Y(I) IN000220
AYP(I)=YP(I) IN000230
SUM1(I)=0. IN000240
6 SUM2(I)=0. IN000250
GO TO 7 IN000260
2 AL1=H/2. IN000270
AL2=H/4. IN000280
AL3=.5 IN000290
AL4=2. IN000300
GO TO 7 IN000310
5 IF (INDT-3) 3,3,4 IN000320
3 X=X+H/2. IN000330
AL1=H IN000340
AL2=H/2. IN000350
AL3=1. IN000360
AL4=2. IN000370
7 DO 8 I=1,NEQ IN000380
IF (I -NSO) 9,9,10 IN000390
Y(I)=AY(I)+AL1*AYP(I)+AL2*FR(I) IN000400
FR(I)=H*F(I) IN000410
YP(I)=AYP(I)+AL3*FR(I) IN000420
SUM1(I)=SUM1(I)+FR(I) IN000430
GO TO 11 IN000440
10 FR(I)=H*F(I) IN000450
Y(I)=AY(I)+AL3*FR(I) IN000460
11 SUM2(I)=SUM2(I)+AL4*FR(I) IN000470
12 CONTINUE IN000480
RETURN IN000490
13 DO 12 I=1,NEQ IN000500
FR(I)=H*F(I) IN000510
SUM2(I)=(SUM2(I)+FR(I))/6. IN000520
IF (I-NSO) 13,13,14 IN000530
13 Y(I)=AY(I)+H*AYP(I)+SUM2(I)/6. IN000540
YP(I)=AYP(I)+SUM2(I) IN000550
14 GO TO 12 IN000560
17 Y(I)=AY(I)+SUM2(I) IN000570
18 CONTINUE IN000580
RETURN IN000590
END IN000600

```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR RUNKUT

```

C   SUBROUTINE LINT(ARG,X,Y,NPT,IP,YI)
C   LAGRANGE INTERPOLATION,2-3,4,5 POINT
C   IP IS DEGREE OF INTERPOLATION
C   NPT IS NUMBER OF POINTS GIVEN IN TABLE
C   DIMENSION X(2),Y(2)
C   T=ARG
C   I=1
20   IF(T-X(1))40,25,30
25   YI=Y(I)
      GO TO 400
30   I=I+1
      IF(I-NPT)20,20,35
35   I=NPT
40   GO TO (40,55,60,60,60),IP
C   2 POINT INTERPOLATION
55   IF(I-1)56,56,57
56   I=I+1
      CONTINUE
      C01=(T-X(I))/((X(I-1)-X(I)))
      C02=(T-X(I-1))/((X(I)-X(I-1)))
      YI=C01*Y(I-1)+C02*T(I)
      GO TO 400
60   IF(I-IP*2)95,95,110
95   I=I+1
      GO TO 60
110  IF(I+1-NPT)145,145,140
140  I=I-1
      GO TO 110
145  GO TO (145,145,100,200,300),IP
C   3 POINT INTERPOLATION
100  C1=X(I-1)
      C2=X(I)
      C3=x(i+1)
      C01=((I-C2)*(I-C3))/((C1-C2)*(C1-C3))
      C02=((I-C1)*(I-C3))/((C2-C1)*(C2-C3))
      C03=((I-C1)*(I-C2))/((C3-C1)*(C3-C2))
      YI=C01*Y(I-1)+C02*Y(I)+C03*Y(I+1)
      GO TO 400
C   4 POINT INTERPOLATION
200  C1=X(I-2)
      C2=X(I-1)
      C3=x(i)
      C4=x(i+1)
      D1=C1-C2
      D2=C1-C3
      D3=C1-C4
      D4=C2-C3
      D5=C2-C4
      D6=C3-C4
      T1=T-C2
      T2=T-C3
      T3=T-C4
      T4=T-C1
      C01=(T1*T2*T3)/(D1*D2*D3)
      C02=-(T4*T2*T3)/(D1*D4*D5)
      C03=(T4*T1*T3)/(D2*D4*D6)
      C04=-(T4*T1*T2)/(D3*D5*D7)
      YI=C01*Y(I-2)+C02*Y(I-1)+C03*Y(I)+C04*Y(I+1)
      GO TO 400
C   5 POINT INTERPOLATION

```

```
300 C1=x(1-3)
C2=x(1-2)
C3=x(1-1)
C4=x(1)
C5=x(1+1)
C1=C1-C2
L2=C1-C3
L3=C1-C4
L4=C1-L5
L5=C2-L3
L6=C2-L4
L7=C2-L5
L8=C3-L4
L9=C3-L5
L10=C4-L5
T1=T-C2
T2=T-C3
T3=T-C4
T4=T-C5
T5=T-C1
C01=-(T1*T2*T3*T4)/(D1*D2*D3*D4)
C02=-(T5*T1*T2*T3)/(D1*D5*D6*D7)
C03=-(T5*T1+T2*T4)/(D2*D5*D8*D9)
C04=-(T5*T1*T2*T4)/(D3*D6*D8*D10)
C05=(T5*T1*T2*T3)/(D4*D7*D9*D10)
Y1=C01*Y(1-3)+C02*Y(1-2)+C03*Y(1-1)+  
    C04*Y(1)+C05*Y(1+1)
400 RETURN
END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR LINT

NO ERRORS

3200 FORTRAN (2.1.0)/(HTS)

```
SUBROUTINE SIMEQ(A,N,C)
DIMENSION A(21,22),B(21,22),C(21)
TOL=5,0E-08
KA=N+1
K2=2
K3=1
DO 10 I=1,N
DO 10 J=1,KA
MOVE MATRIX TO WORKING AREA
10 B(I,J)=A(I,J)
20 IF(ABS(B(K3,K3))=TOL)30,100,100
30 IF(K2=N)40,40,300
40 DO 50 I=1,KA
D=B(K3,I)
B(K3,I)=B(K2,I)
B(K2,I)=D
50 CONTINUE
K2=K2+1
GO TO 20
100 K1=K3
K3=K3+1
K2=K3+1
IF(K3=N)120,120,200
120 DO 130 I=K3,KA
B(K1,I)=B(K1,I)/B(K1,K1)
130 CONTINUE
DO 140 I=K3,N
DO 140 J=K3,KA
B(I,J)=B(I,J)-B(I,K1)*B(K1,J)
140 CONTINUE
GO TO 20
200 C(N)=B(N,N+1)/B(N,N)
K3=N-1
210 K1=K3+1
L=0,
DO 220 I=K1,N
220 L=L+B(K3,I)*C(I)
C(K3)=B(K3,KA)-L
K3=K3+1
IF(K3)350,350,210
300 TOL=TOL/10.
IF(TOL-1.0E-30)320,310,310
310 K2=K3+1
GO TO 20
320 WRITE(61,321)
321 FORMAT(/23H NO SOLUTION FROM SIMEQ//)
350 RETURN
END
```

3200 FORTRAN DIAGNOSTIC RESULTS - FOR SIMEQ

NO ERRORS

## REFERENCES

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TABLE I. - INITIAL CONDITIONS FOR THREE-STREAM MIXING

Physical Property	LOCATION		
	Jet Exit	Turbine Exhaust Exit	Free Stream
Temperature, °K	2751.5	830.0	281.1
Pressure, dyne/cm <sup>2</sup>	$8.9149 \times 10^5$	$8.9149 \times 10^5$	$8.9149 \times 10^5$
Density, gm/cm <sup>3</sup>	$9.10712 \times 10^{-5}$	$1.17099 \times 10^{-4}$	$1.1005 \times 10^{-3}$
Speed, cm/sec	293,126.1	36,576.0	9,144.0
Enthalpy, cal/gm	73.38	-1,049.25	125.34
Mass Fractions			
c <sub>H<sub>2</sub></sub>	0.04377	0.16558	0.0
c <sub>O<sub>2</sub></sub>	0.0	0.0	0.233
c <sub>H<sub>2</sub>O</sub>	0.02246	0.0	0.0
c <sub>C</sub>	0.0	0.10093	0.0
c <sub>CO</sub>	0.00428	0.0	0.0
c <sub>CO<sub>2</sub></sub>	0.93267	0.67101	0.0
c <sub>CH<sub>4</sub></sub>	0.0	0.06249	0.0
c <sub>N<sub>2</sub></sub>	0.0	0.0	0.767

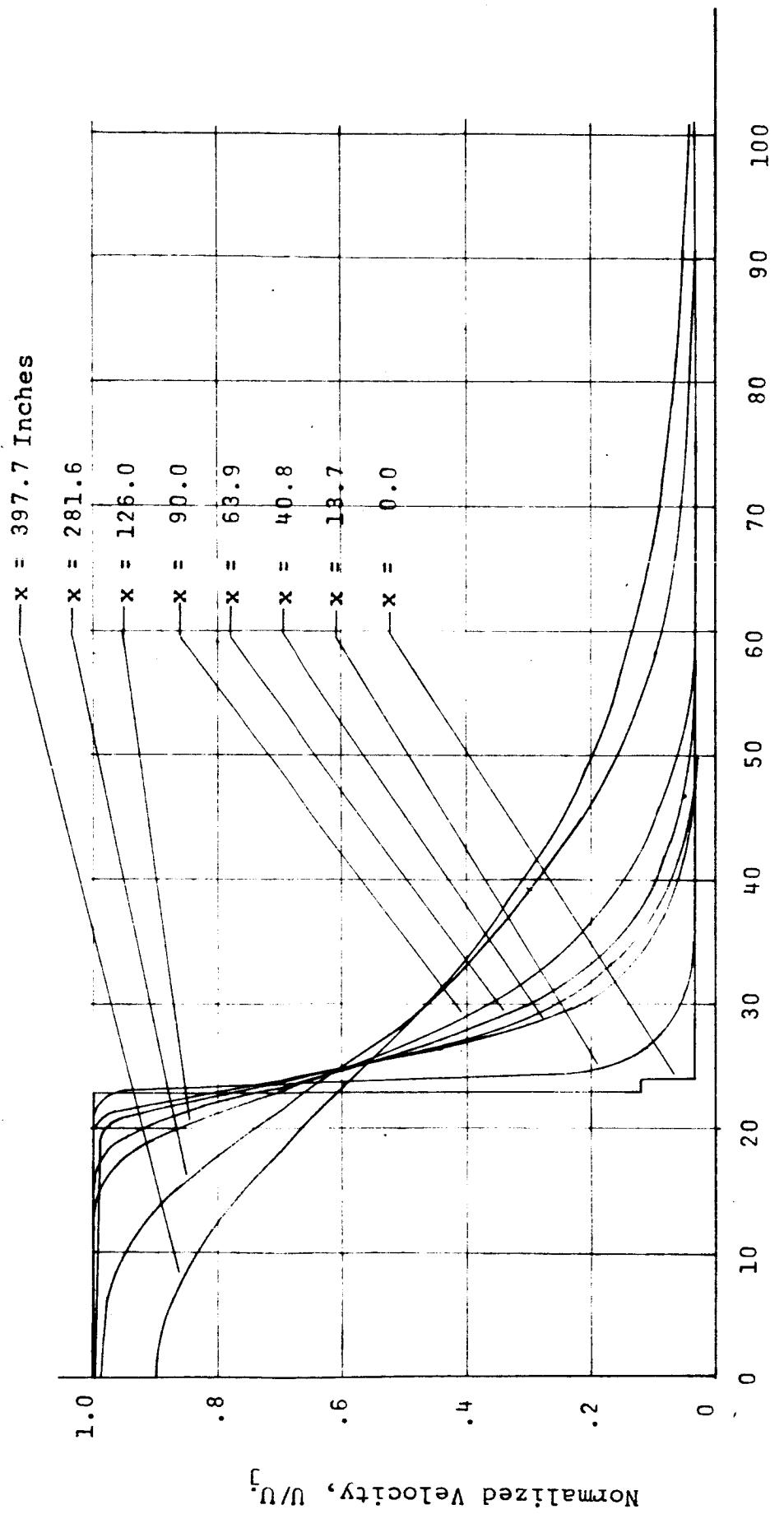


FIG. 1 VELOCITY PROFILE FOR THREE-STREAM-MIXING

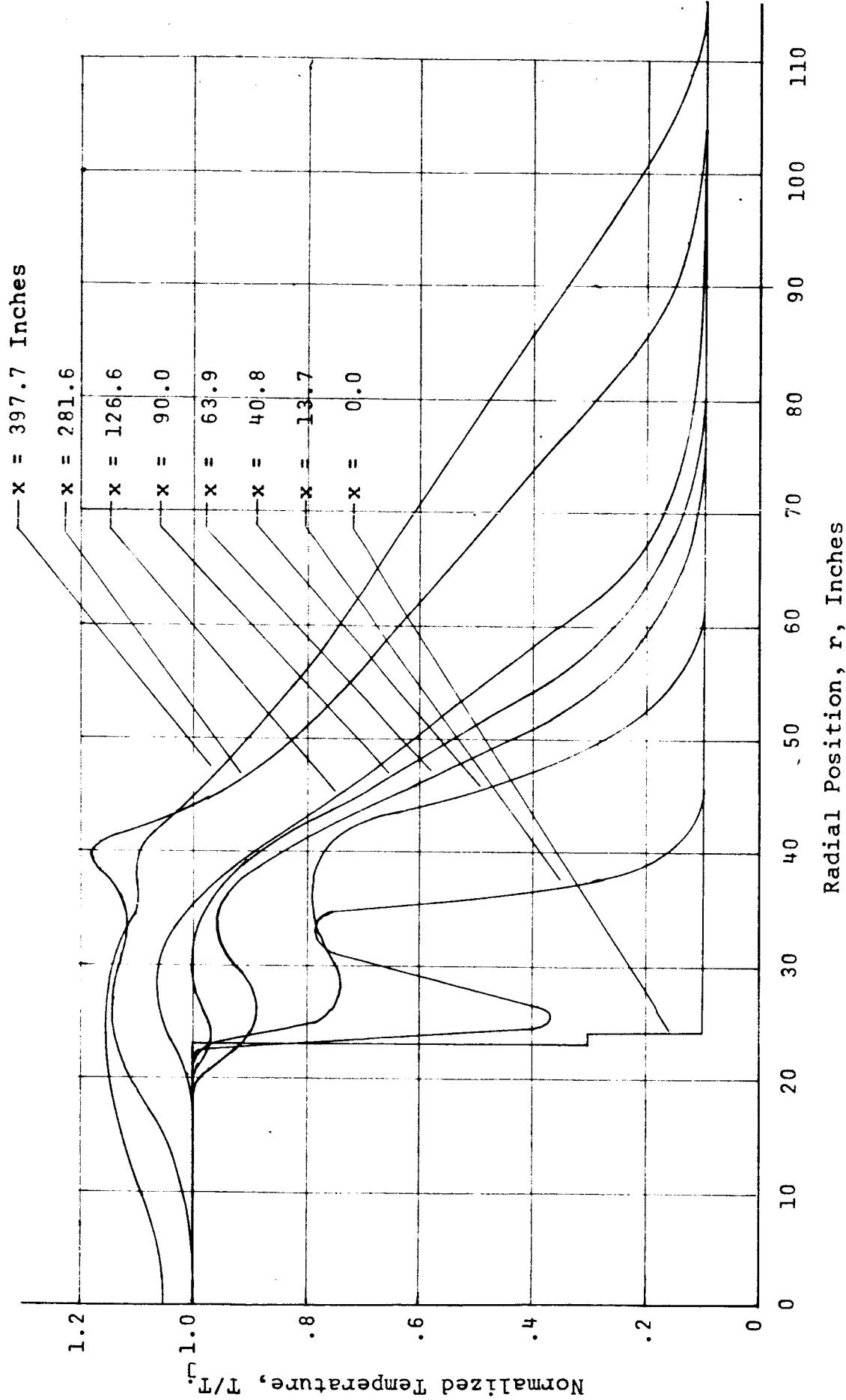


FIG. 2 TEMPERATURE PROFILE FOR THREE-STREAM-MIXING

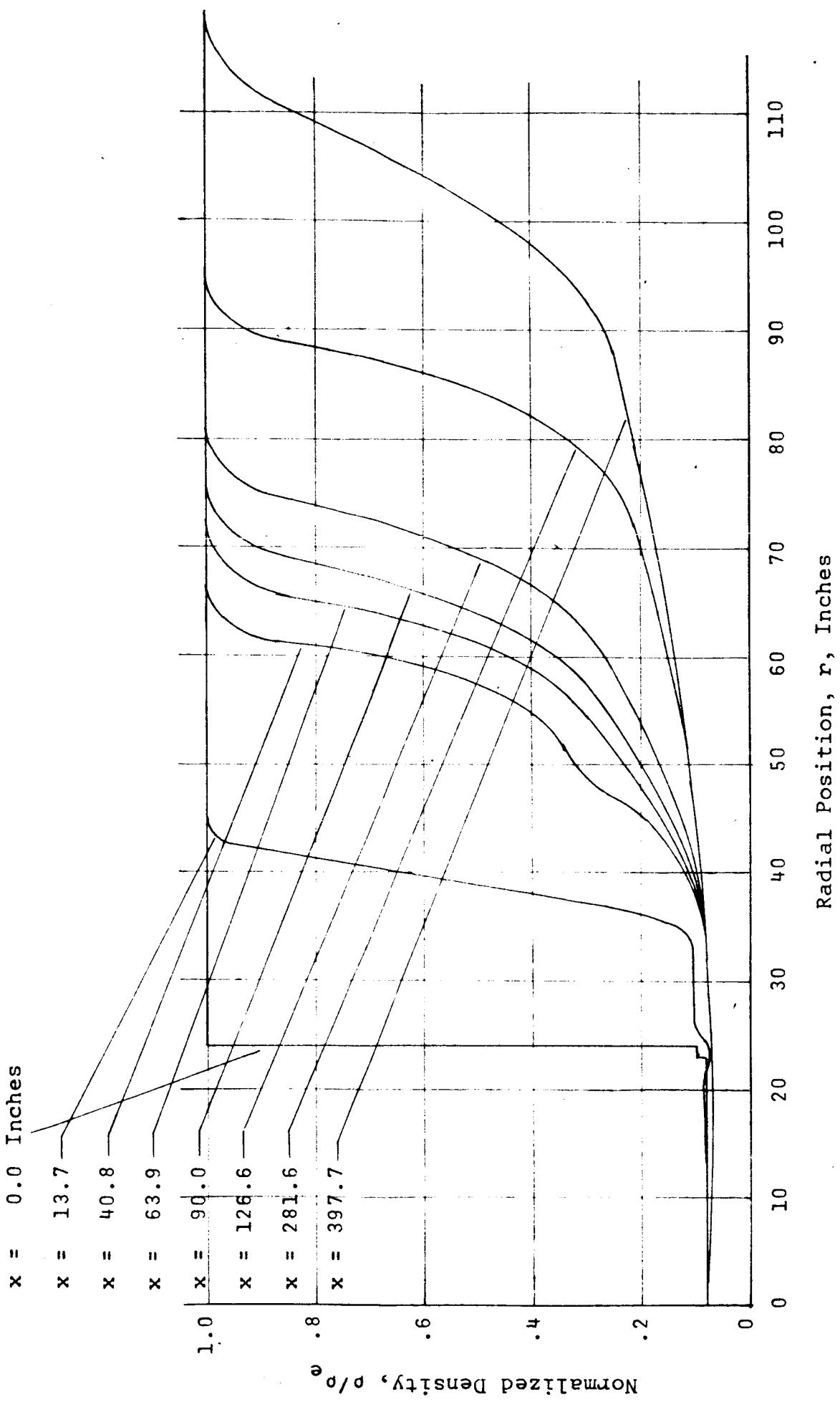


FIG. 3 DENSITY PROFILE FOR THREE-STREAM-MIXING

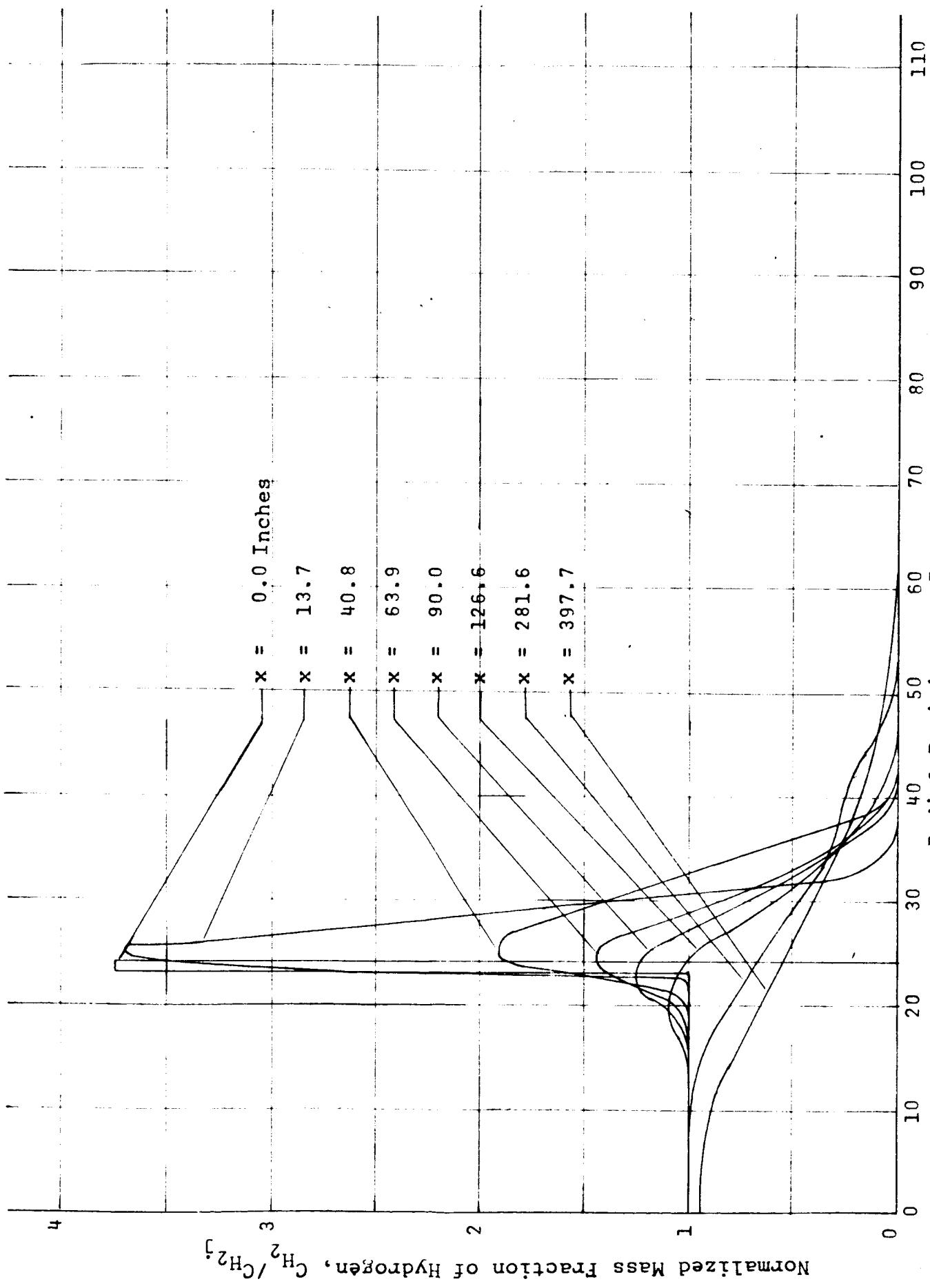


FIG. 4 MASS FRACTION OF HYDROGEN PROFILE FOR THREE-STREAM-MIXING

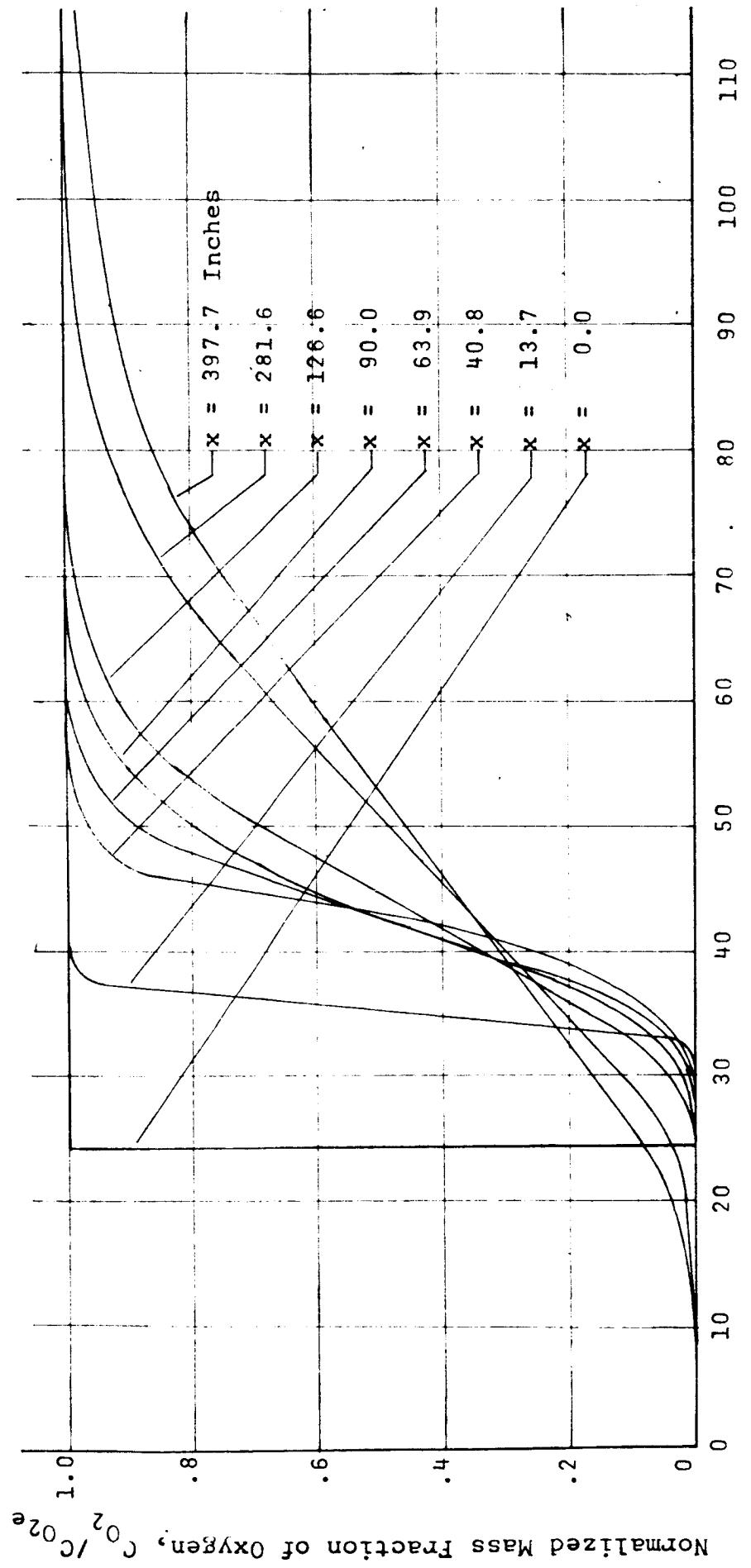


FIG. 5 MASS FRACTION OF OXYGEN PROFILE FOR THREE-STREAM-MIXING

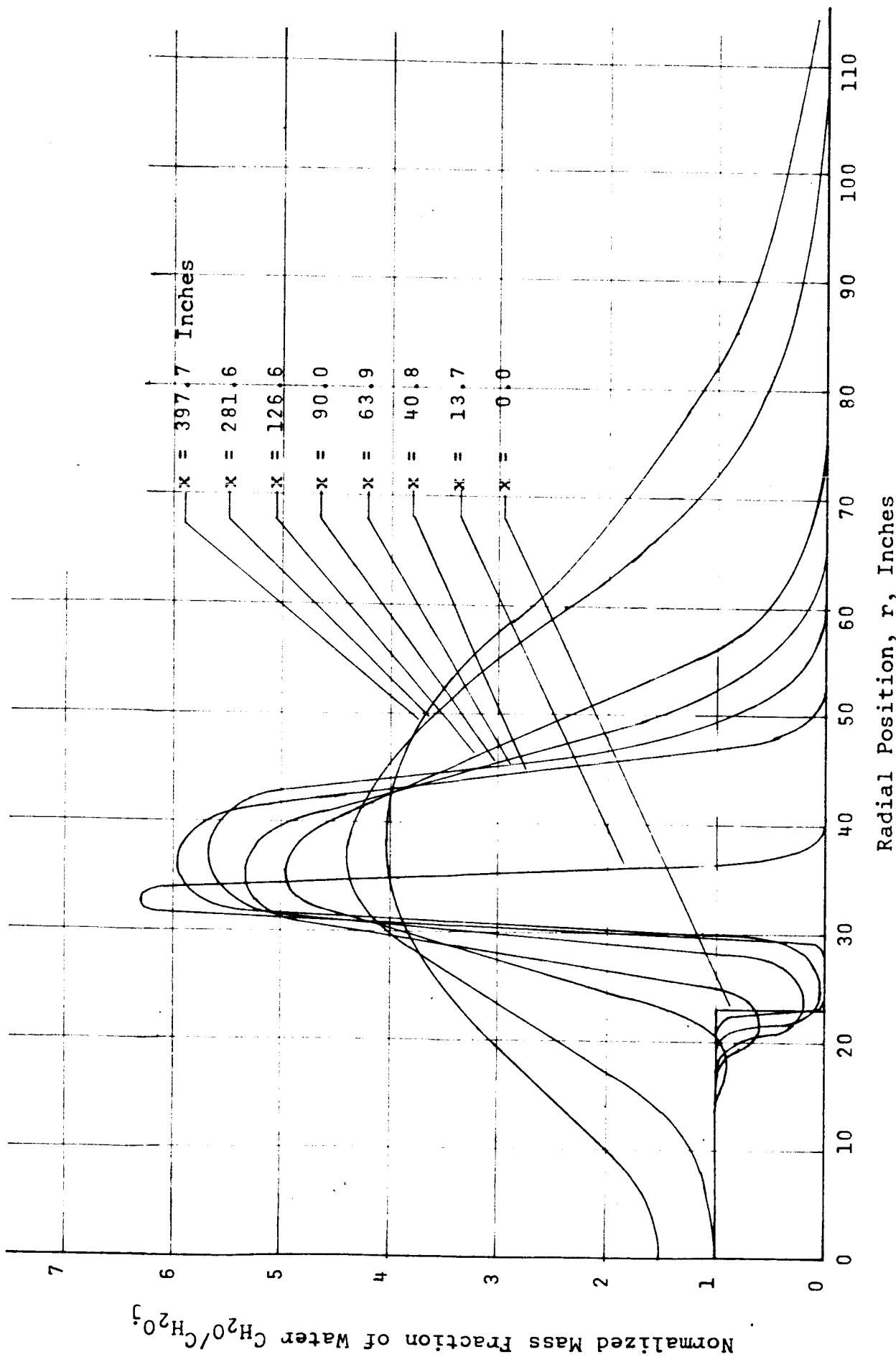


FIG. 6 MASS FRACTION OF WATER PROFILE FOR THREE-STREAM-MIXING

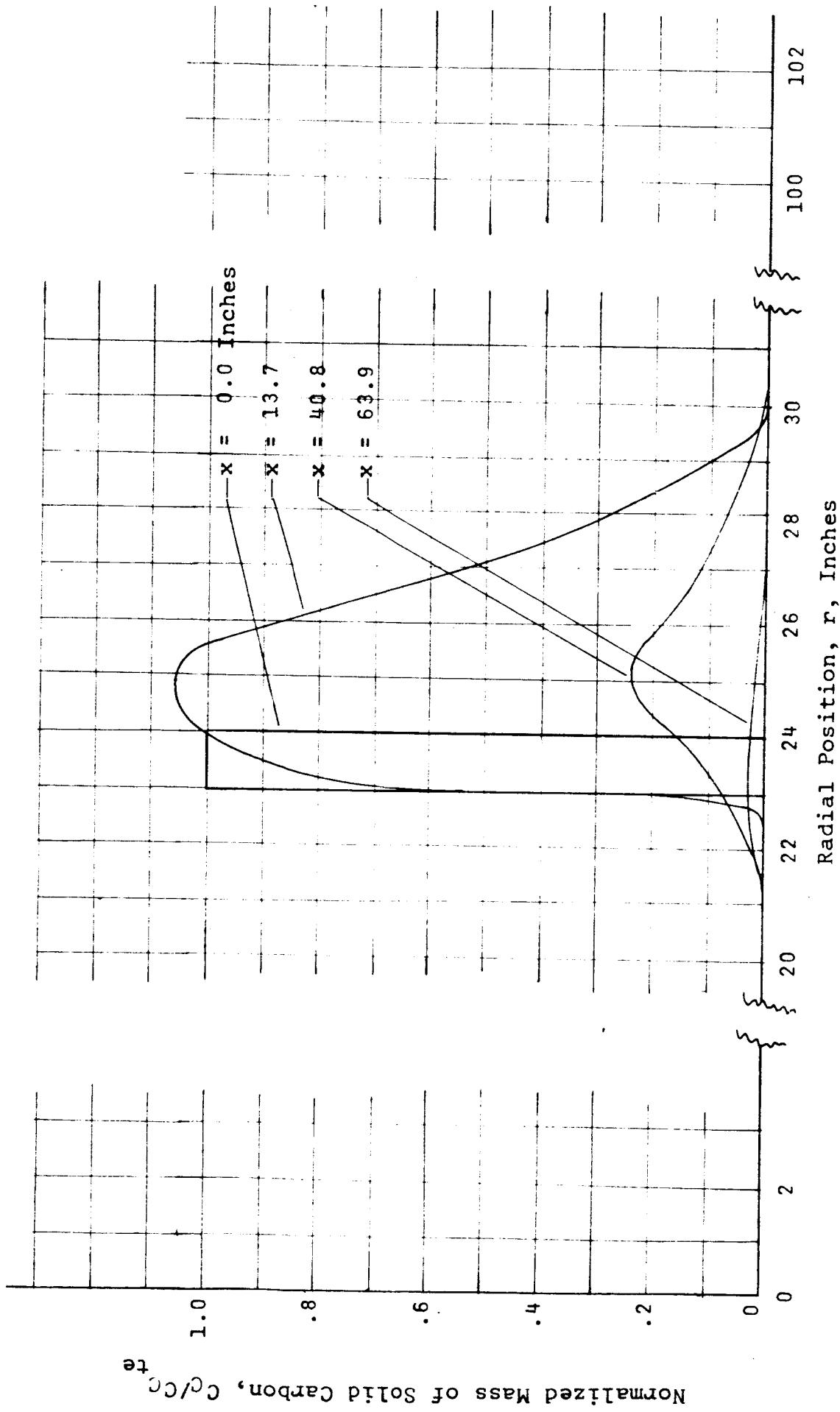


FIG. 7 MASS FRACTION OF SOLID CARBON PROFILE FOR THREE-STREAM-MIXING

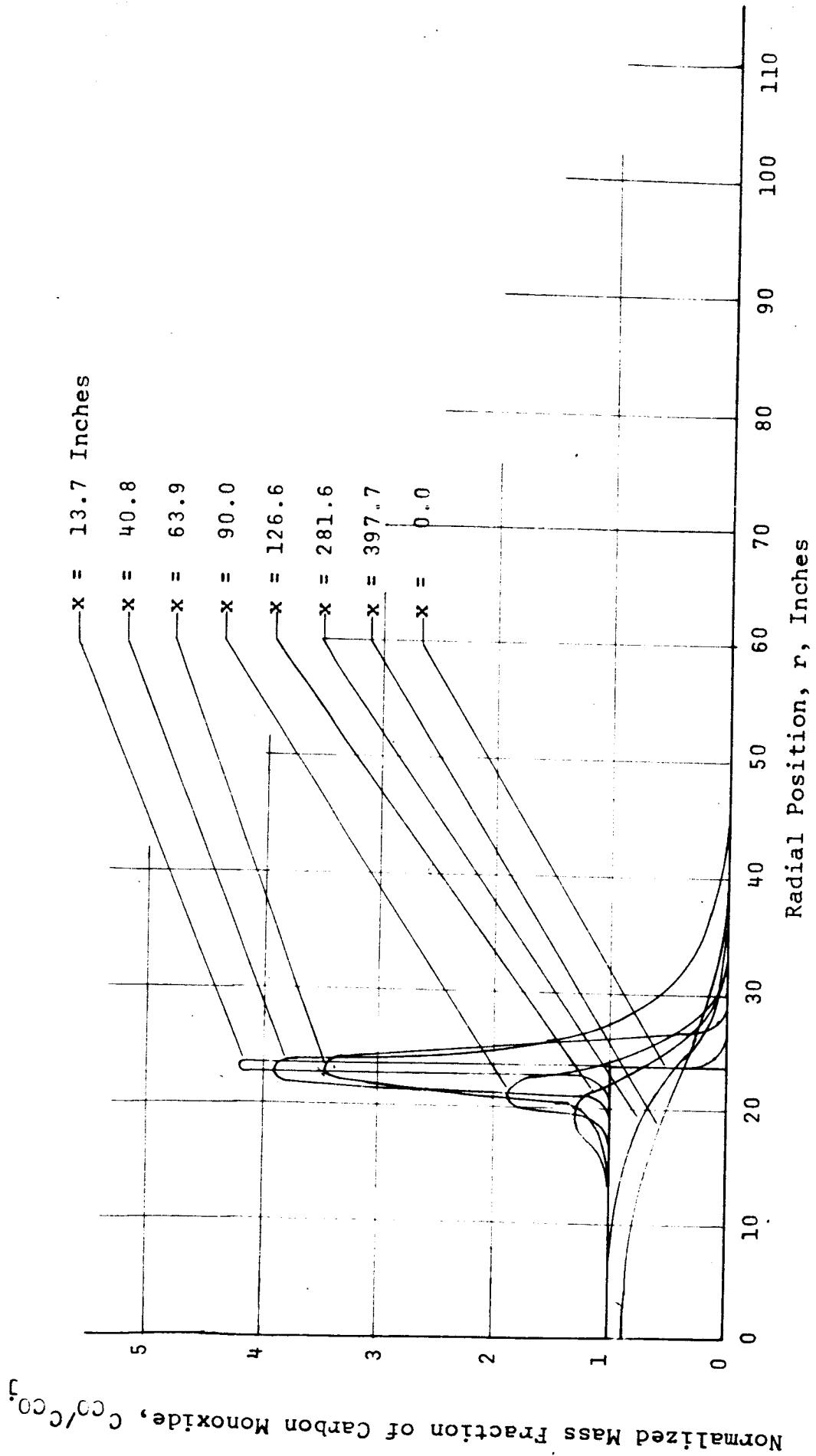


FIG. 8 MASS FRACTION OF CARBON MONOXIDE PROFILE FOR THREE-STREAM-MIXING

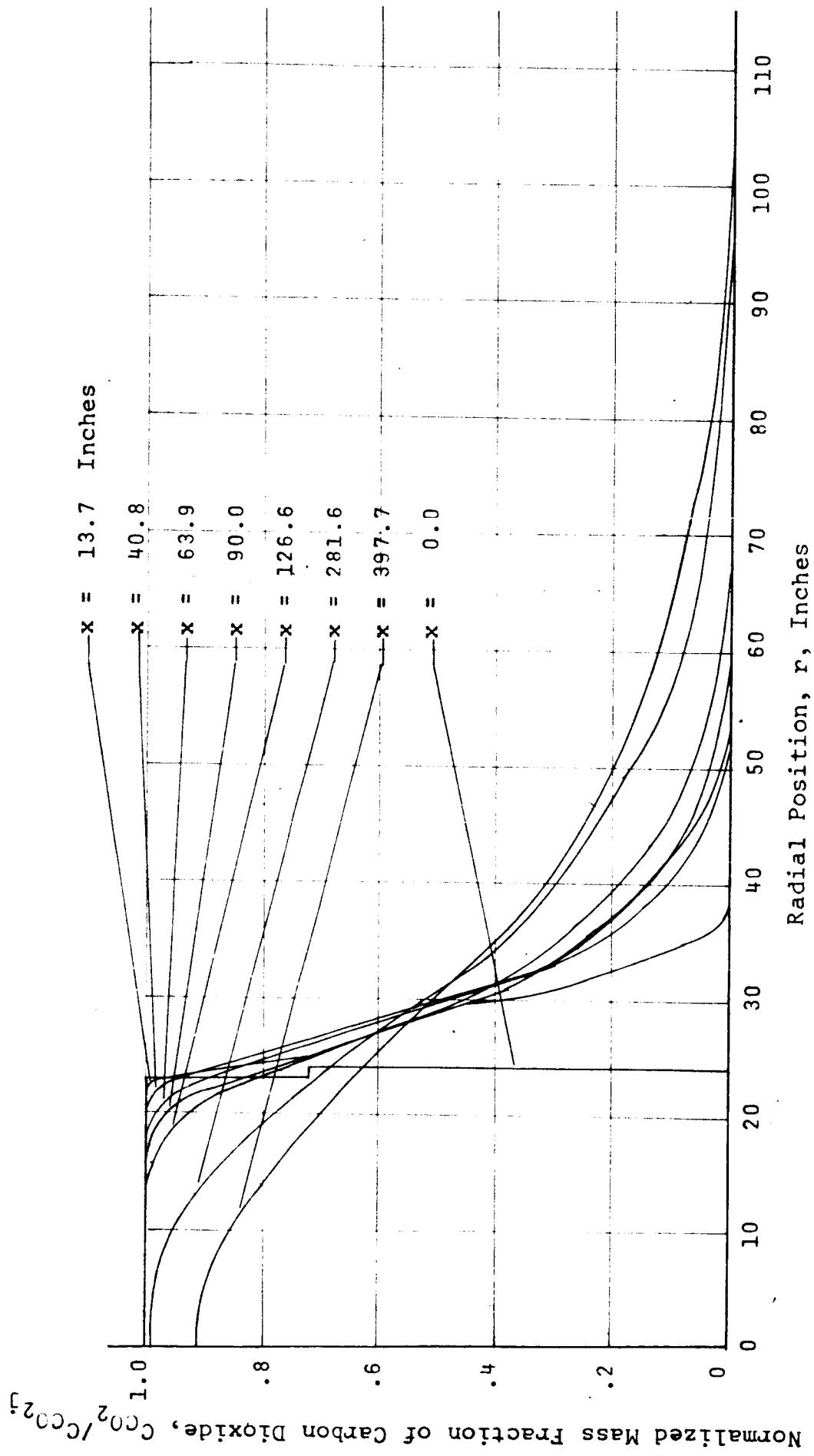


FIG. 9 MASS FRACTION OF CARBON DIOXIDE PROFILE FOR THREE-STREAM-MIXING

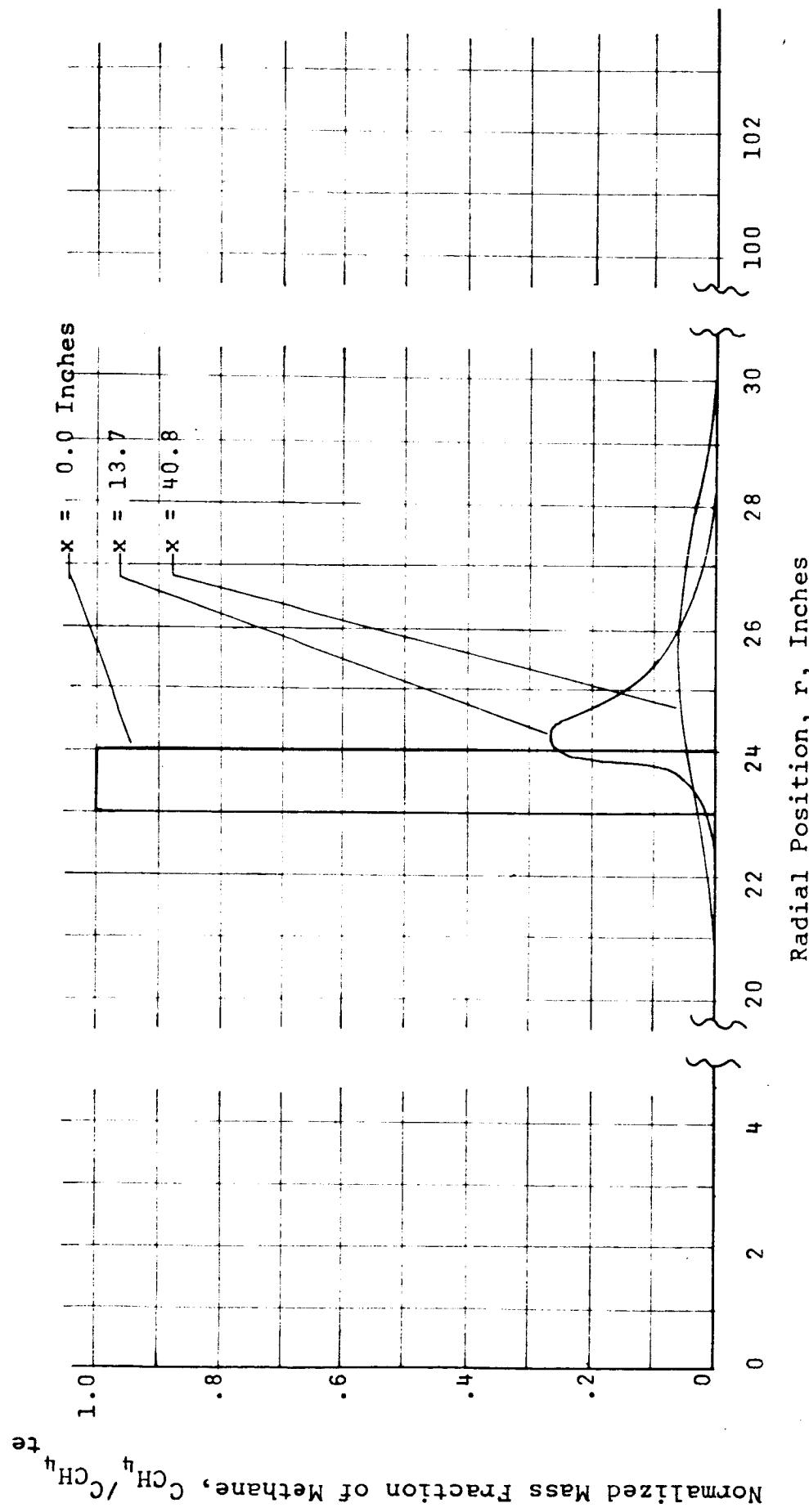


FIG. 10 MASS FRACTION OF METHANE PROFILE FOR THREE-STREAM-MIXING

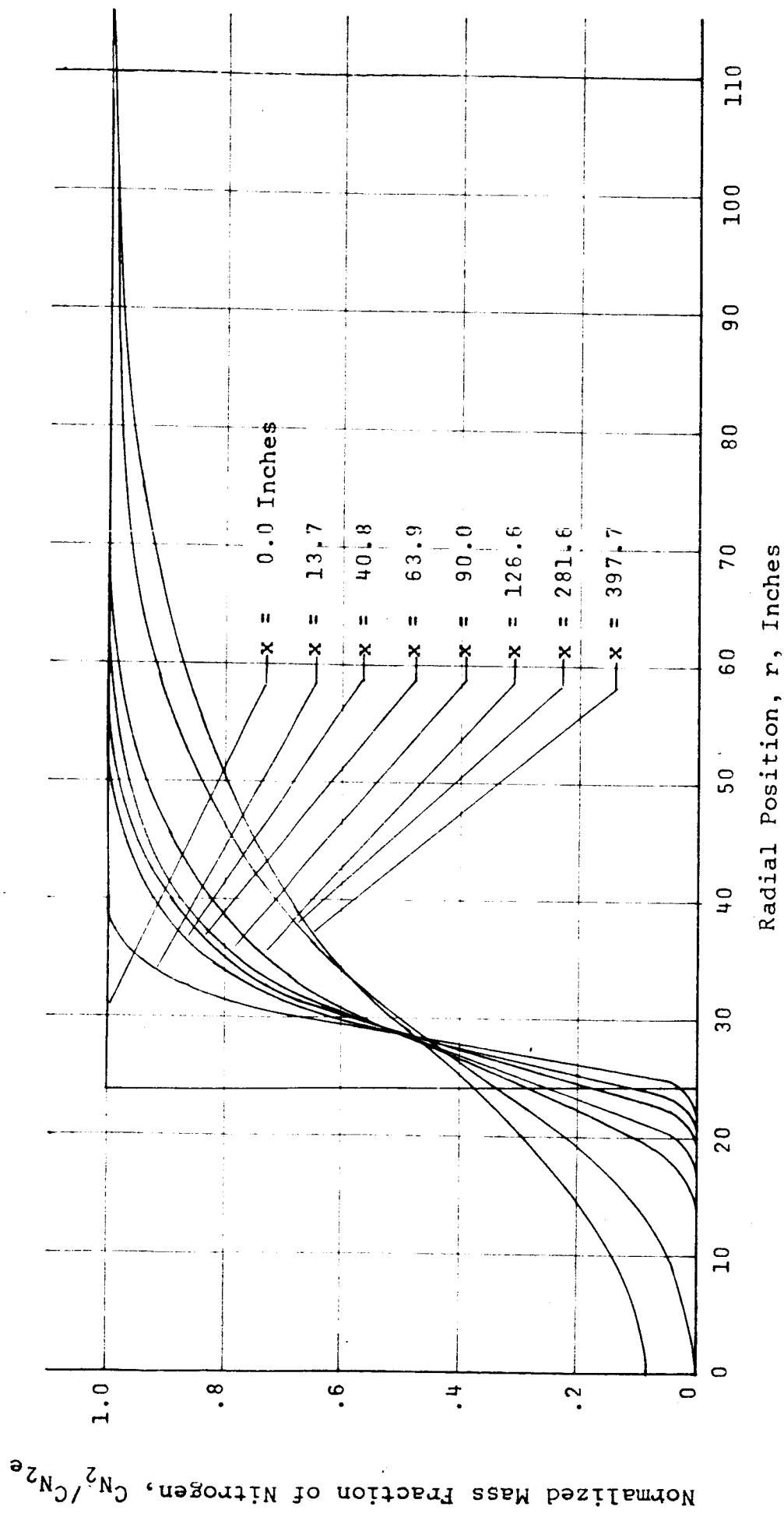


FIG. 11 MASS FRACTION OF NITROGEN PROFILE FOR THREE-STREAM-MIXING

NRO=4 NSMAX=7 NRS=3  
NRO1=5 NSMAX1=8 NRS1=4  
NRS=NRS=NSMAX1—NRO1

OPTIONAL FIELD DESIGN

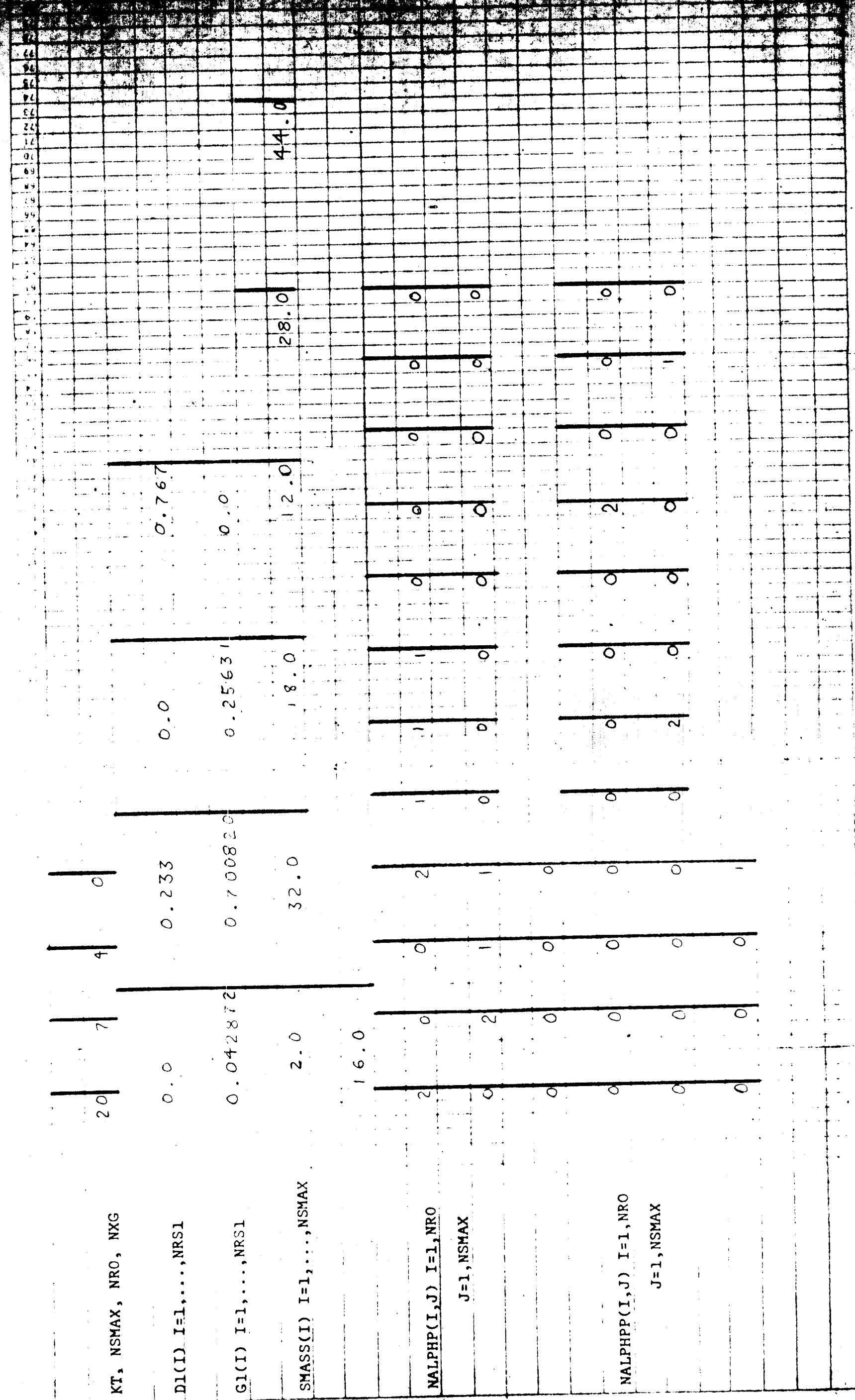


FIG. 12 TWO-STREAM INPUT

## GENERAL CARD DESIGN

PSI, DPSI, DPSIBAR, XI, 0.0	0 - 0.5	0.01	- 0.025	0.005	1 - 1.4
DXI, PSIMAX	281 - 1.1	1.100505E-3	0.034670	125.3393436	73.382
XIMAX, TREF, PREF, REFL, 0.0001	1 - 0	2600.0	100.0	100.0	10.0
HE, HJ, ACCTOL, PSIJ, T, DELTAT, 10.0	2931 - 26.1	9144.0	0.0	A.1071203E-5	24.28401
DT, TOL	28.0				
PSIBAR, UJ, UE, USTARJ, 0.0					
RHOE, RJ					
CNITROM					
AACON (I,J,1)	3.0798700	5.421663E-4	3.848762E-8	-3.46338E-11	3.6791E-15
I, =1,6 J=1, NSMAX	3.5980995	7.805671E-4	-2.233751E-7	4.23966E-11	-3.34191E-15
(Higher Temperature Range)	2.6707532	3.031711E-3	-8.535157E-7	1.17908E-10	-6.19735E-15
1.35222089	1.878796E-3	-7.822261E-7	1.553577E-10	-1.18144E-14	-645.83539
2.9537179	1.547681E-3	-6.160384E-7	1.12728E-10	-7.72089E-15	-1.423281E+4
4.4008955	3.216385E-3	-1.313466E-6	2.45482E-10	-1.70946E-14	-4.894009E+4
1.1795744	1.095059E-2	-4.062213E-6	7.13703E-10	-4.74904E-14	-9.855663E+3
AACON (I,J,2)	2.8612126	4.063282E-3	-9.192602E-6	8.94899E-09	-3.05132E-12
I=1,6 J=1, NSMAX	3.7189946	-2.516728E-3	8.583935E-6	-8.29987E-09	2.70822E-12
(Lower Temperature Range)	4.1565016	-1.724433E-3	5.698231E-6	-4.59300E-09	1.42336E-12
-6.4250569	6.826247E-3	-4.225572E-6	1.522231E-10	4.81920E-13	-75.029564
3.7929971	-2.215637E-3	5.193912E-6	-3.603526E-9	8.22622E-13	-1.436408E+4
2.1974487	1.016306E-2	-1.016489E-5	5.724671E-9	-1.38808E-12	-4.835513E+4
4.2497878	-6.912656E-3	33.160213E-5	-2.971543E-8	9.51036E-12	-1.018663E+4
AAHOT (I,J) I=1,6	1.585176E-3	-6.177990E-7	1.11194E-10	-7.52106E-15	-8.924897E+2
J=1,2	3.6927389	-1.341199E-3	2.669338E-6	-9.95596E-10	-9.37266E-14
Aθ, PLYC(I) I=1,20	1.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0
	0.0	0.0	0.0	0.0	0.0

FIG. 13 TWO-STREAM INPUT-B

## GENERAL CARD DESIGN

TTAB(K) K=1, KT	600.0	800.0	1000.0	1200.0	1400.0	1600.0	1800.0	2000.0	2200.0	2400.0	2600.0	2800.0	3000.0	3200.0	3400.0	3600.0	3800.0	4000.0
STAB(I,J) I=1, KT	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
J=1, NSMAX	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
1	1.942	2.443	2.021	1.658	1.343	1.067	0.842	0.607	0.413	0.238	0.079	0.0	0.0	0.0	0.0	0.0	0.0	0.0
2	0.842	0.607	0.413	0.238	0.079	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
8	0.771	8.234	7.81	7.469	7.185	6.946	6.741	6.563	6.407	6.269	6.145	6.034	5.933	5.841	5.756	5.679	5.405	5.30
9	0.411	8.625	7.960	7.388	6.892	6.458	6.074	5.732	5.425	5.149	4.898	4.670	4.001	-1.801	-2.372	-2.803	-1.801	-1.801

## GENERAL CARD DESIGN

-3.139	3.408	-3.627	-3.809	-3.962	-4.093
-4.206	-4.304	-4.391	-4.467	-4.536	-4.598
-4.653	-4.704				
KSPEC (I,J) I=1,4 J=1, NSMAX	4	0	0	0	0
	0	0	0	0	0
XG(I) I=1, ..., NSMAX	0.011943	0.0000548	0.27855	1.0	E-050.40266
XG(I) CUT-OFF	1.0	E-05			
SUM	0.0	E-40			

**NRO = 4 NSMAX = 7 NRS = 3**  
**NRO1 = 5 NSMAX1 = 8 NRS1 = 4**  
**NRS=NRS=NSMAX1—NRO1**

## GENERAL CARD DESIGN

Kt, NSMAX, NRO, NXG

D1(I) I=1,...,NRS1

G1(I) I=1,...,NRS1

SMASS(I) I=1,...,NSMAX

NALPHPP(I,J) I=1,NRO  
J=1,NSMAX

FIG. 16 THREE-STREAM INPUT-A

## GENERAL CARD DESIGN

PSI, DPSI, DPSIBAR, XI, 0.0	0.05	0.025	0.005	1.4
XIMAX, TREF, PREF, REFLV, 0.0001	281.11	1.103505E-3	0.034670	125.3393436 73.382
HE, HJ, PSIJ, T, DELTAT, 10.0	1.0	2600.0	100.0	10.0
ACCTOL, PSIJ, DT, TOL, 0.0	2931.26.1	9144.0	0.0	A.1071203E-5 24.28401
PSIBAR, UJ, UE, USTARJ, 0.0	28.0			
RHOE, RJ				
CNITROM				
AACON (I,J,1)	3.0798700	5.421663E-4	3.848762E-8	-3.46338E-17 3.67191E-15 -8.689205E+2
I, =1,6 J=1, NSMAX	3.5980995	7.805671E-4	-2.233751E-7	4.23966E-17 3.34191E-15 -1.192975E+3
(Higher Temperature Range)	2.6707532	3.031711E-3	-8.535157E-7	1.17908E-10 -6.19735E-15 -2.9888999E+4
1.35222089	1.878796E-3	-7.822261E-7	5.53577E-10	-1.18144E-14 -6.451.83539
2.9537179	1.547681E-3	-6.160384E-7	1.12728E-10	-7.72089E-15 -1.423281E+4
4.4008955	3.216385E-3	-1.313466E-6	2.45482E-10	-1.70946E-14 -4.894009E+4
1.1795744	1.095059E-2	-4.062213E-6	7.13703E-10	-4.74904E-14 -9.855663E+3
AACON (I,J,2)	2.8612126	4.063282E-3	-9.192602E-6	8.94899E-09 -3.05132E-12 -9.686986E+2
I=1,6 J=1, NSMAX	3.7189946	-2.516728E-3	8.583935E-6	-8.29987E-09 2.70822E-12 -1.057671E+3
(Lower Temperature Range)	4.1565016	-1.724433E-3	5.698231E-6	-4.59300E-09 1.42336E-12 -3.028877E+4
-64250569	6.826247E-3	-4.225572E-6	1.522231E-10	4.81929E-13 -75.029564
3.7929971	-2.215637E-3	5.193912E-6	-3.603526E-9	8.22622E-13 -1.436408E+4
2.1974487	1.016306E-2	1.016489E-5	5.724671E-9	-1.38808E-12 -4.835513E+4
4.2497878	-6.912656E-3	-2.971543E-5	9.51036E-12	-1.018663E+4
1.585176E-3	-6.177990E-7	1.11194E-10	-7.52106E-15	-8.924897E+2
2.8609546				
J=1,2	3.6927389	-1.341199E-3	2.669338E-6	-4.95596E-10 -9.37266E-14 -1.062950E+3
AAHOT (I,J) I=1,6				

FIG. 17 THREE-STREAM INPUT-B

## GENERAL CARD DESIGN

PSI TABLE	0.0	0.938129	0.942866	0.943818	0.944790	0.945856
	0.947213	0.949147	0.951969	0.955824	0.961346	0.961796
	0.963856	0.965916	0.967975	0.970035	0.972095	0.974154
	0.976214	0.978273	0.980333	0.982392	0.984452	0.986743
	0.987445	0.988169	0.988891	0.989742	0.990862	0.992197
	0.993657	0.994900	0.995807	0.996533	0.997130	0.997703
	0.998277	0.998851	0.999429	1.0	0.99788	0.984651
MODIFICATION FACTOR	1.0	1.0	0.99981	0.99754	0.111893	0.92887
FOR VELOCITY	0.78326	0.54804	0.312946	0.16754	0.096492	
	0.096517	0.096534	0.096546	0.096555	0.096562	0.0965674
	0.0965716	0.096575	0.096718	0.096580	0.096474	0.090383
	0.084229	0.074757	0.062155	0.047666	0.03326	0.020894
	0.011709	0.005815	0.002546	0.000979	0.000329	0.000097
	0.000025	0.000006	0.000001	0.0	0.0	
MODIFICATION FACTOR	1.0	1.0	1.000653	1.007557	1.054814	1.254035
FOR FIRST ELEMENT	1.774118	2.614177	3.453863	3.973176	4.171930	4.221940
	4.223036	4.223759	4.224296	4.224692	4.224972	4.225205
	4.225392	4.225532	4.225672	4.225742	4.221123	3.954632
	3.685366	3.270923	2.719561	2.085557	1.455449	0.914196
	0.512320	0.256058	0.111385	0.042813	0.014404	0.004231
	0.001082	0.000240	0.000047	0.0	0.0	
MODIFICATION FACTOR	1.0	1.0	0.999906	0.998936	0.992273	0.964196
FOR SECOND ELEMENT	0.890860	0.772421	0.654038	0.580820	0.552798	0.544504
	0.544643	0.543978	0.544808	0.544857	0.544925	0.544925
	0.544951	0.544968	0.544983	0.544996	0.544400	0.510030

## GENERAL CARD DESIGN

0	475302	0.421850	0.350740	0.268975	0.187711	0.117904
0	066075	0.032814	0.014364	0.005521	0.001858	0.000545
0	000139	0.000032	0.000006	0.0		
MODIFICATION FACTOR	1.0	1.000078	1.00063	1.004939	1.022886	
FOR THIRD ELEMENT	1.069736	1.145414	1.221057	1.267840	1.285744	1.289259
	1.289591	1.289817	1.289938	1.290098	1.290138	1.290258
	1.290316	1.290359	1.290398	1.290426	1.289013	1.207635
	1.125407	0.99885	0.830475	0.636869	0.444458	0.279170
	0.156420	0.077695	0.0340139	0.013074	0.004399	0.001292
	0.000331	0.000074	0.000014	0.0		
MODIFICATION FACTOR	1.0	1.0	1.0	1.0	1.0	1.0
FOR FOURTH ELEMENT	1.0	1.0	1.0	1.0	1.0	1.0
	1.0	1.0	1.0	1.0	1.0	1.0
	0.87196	0.77390	0.64345	0.49345	0.34436	0.21630
	0.121218	0.06020	0.02635	0.010130	0.003408	0.001002
	0.000258	0.000057	0.0	0.0	0.0	
MODIFICATION FACTOR	1.0	1.0	1.0	1.0	1.0	1.0
FOR TOTAL ENTHALPY	6.183883	11.809516	17.432454	20.910176	22.241122	22.582366
	22.588179	22.592143	22.594973	22.597071	22.598649	22.599881
	22.600862	22.601651	22.602306	22.602787	22.578036	20.459765
	19.712397	17.495564	14.546394	11.155282	7.784995	4.889851
	2.740304	1.360875	0.595781	0.228997	0.077044	0.022634
	0.005793	0.001290	0.000250	0.0		

FIG. 19. THREE-STREAM INPUT

GENERAL CARD DESIGN

Nest C - Figure 103 (July 1980)

FIG. 21 THREE-STREAM INPUT-F

CARDS DESIGN

TTAB(K) K=1, KT	600.0	800.0	1000.0	1200.0	1400.0	1600.0
1 800.0	2000.0	2200.0	2400.0	2600.0	2800.0	3000.0
3 000.0	3200.0	3400.0	3600.0	3800.0	4000.0	
4 200.0	4400.0					
STAB(I,J) I=1, KT	0.0	0.0	0.0	0.0	0.0	0.0
J=1, NSMAX	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
1 0.062	7.899	2.443	0.607	0.43	0.238	0.079
2 942	0.842	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
0.0	0.0	0.0	0.0	0.0	0.0	0.0
4.318	8.234	6.563	6.407	7.8	7.469	7.185
7.469	6.269	5.746	5.679	6.407	6.269	6.145
6.145	5.679	5.841	5.746	5.679	5.406	5.034
5.034	5.406	17.243	14.85	12.940	11.504	10.353
6.034	5.406	20.680	17.243	14.85	12.940	11.504
6.946	7.388	7.960	8.625	7.960	7.388	6.892
6.892	5.149	5.425	5.732	5.149	5.425	4.898
4.670	-1.801	-1.211	0.146	0.146	-1.211	-2.372
4.670	-2.803	-2.372	-1.801	-1.211	-0.146	0.001

**FIG. 20 THREE-STREAM INPUT-E**